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DESCRIPTION OF FORTRAN PROGRAM DAWNA FOR  
ANALYSIS OF MUZZLE BLAST FIELD

Prepared by

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SECTION I  
INTRODUCTION

The FORTRAN IV program DAWNA solves the set of partial differential equations governing the flow on the axis of symmetry between the blast wave and the Mach disc of a muzzle blast field. A complete description of the method of solution of the governing equations and statement of the boundary conditions is given in Reference (1). Program DAWNA is a revised edition of program DAWN previously described in Reference (2). The two main refinements included in program DAWNA are the following:

- 1) An initialization technique has been developed that allows the computer program to self-start without the use of empirical relationships (such as the initial locations of the flow discontinuities).
- 2) An acoustic analysis which enables the continuation of the blast field calculation to very late times.

Besides the major analytical refinements presented above, various other modifications have been included in the present computer code which minimize possible sources of numerical error as well as making the program more convenient for the user. The more pertinent of these include the use of automatic grid subdivision, a streamline trace for the determination of the properties upstream of the Mach disc and the option of choosing the units of both the input and output parameters.

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SECTION II  
PROGRAM STRUCTURE

The main program acts as an executive routine which calls the principal subroutines, as follows. Subroutine INDATA is called once to define the finite difference grid and initialize the dependent variables. Subroutines SHOCK, CONTACT and INT PT are called sequentially to advance the solution from time  $t$  to time  $t+\Delta t$ . The dependent variables are then reinitialized, printed and the above sequence repeated. This loop is terminated when the selected number of time steps have been completed, at which time the final solution can be written out in TAPE12 (at the option of the user) and the program execution is stopped.

Subroutines SHOCK and CONTACT accomplish the solution at the two shocks and the contact surface by the method of characteristic technique described in Reference (1). The three surfaces of discontinuity are identified by the indices 1, 2 and 3 which refer to the Mach disc, contact surface and blast wave, respectively. Subroutine INT PT accomplishes the solution at the interior grid points by the finite difference algorithm devised by MacCormack (Reference 3). Within subroutine INT PT the index  $K=1$  denotes the solution between the Mach disc and contact and  $K=2$  denotes the solution between the contact and the blast wave. The indices  $L00P=0$  and 1 refer to the first and second iterates of the MacCormack algorithm.

The subroutines associated with the acoustic wave computation are called from subroutine ACOUS which only returns control to the main program at the completion of the run.

The functions of all the subroutines are summarized in Table I. The main program variables are identified in Table II.

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TABLE I  
SUBROUTINES AND FUNCTIONS

<u>NAME</u>	<u>DESCRIPTION</u>
1. CL	Calculates flow properties on the centerline of the supersonic jet exhaust plume from the muzzle exit conditions and the centerline Mach number at the location of the Mach disc.
2. CONTACT	Calculates the position and velocity of the contact surface and the flow properties on both sides of the contact surface.
3. FS	Sets the flow variables on the upstream side of the blast wave to the specified ambient (i.e., "infinity") conditions.
4. INDATA	Reads the input data and defines accordingly the finite difference grid and the initial values of the dependent variables.
5. INT PT	Computes the new finite difference solution at the interior grid points and the new location of the grid points at each succeeding time step.
6. LPINT	Locates the intersection of a characteristic surface and a time plane, and performs the necessary interpolations of variables from the solution at the grid points.
7. MUZZLE	Determines the pressure, speed of sound and Mach number at the muzzle exit as a function of time from a spline-fit of the corresponding tabular input data.

<u>NAME</u>	<u>DESCRIPTION</u>
8. PUNCH	Writes the final time step on TAPE12 for restarting the calculation (see Section III(B) for an explanation of the restart capability of the program).
9. SETN	Reinitializes the arrays in which the new solution at time $t+\Delta t$ will be stored.
10. SHOCK	Calculates the position and velocity of a moving shock and the flow properties on the downstream side of the shock. The index K in the calling sequence is used to denote the Mach disc (K=1) and the blast wave (K=3).
11. SPLINE	Performs a spline-fit of tabular data (see Reference 4).
12. SPLINT	Uses the spline-fit coefficients to interpolate data at arbitrary values of the independent argument. (First and second derivatives of the dependent variable are also calculated, but not used in the present program.)
13. STEP	Computes the maximum allowable time step, $\Delta t$ , based on the Courant, Friedrichs and Lewy criterion.
14. INTER	Interpolates data when adding or deleting a grid point.
15. INIT	Determines the initial location and fluid properties of the Mach disc, contact surface and blast wave.
16. ACOUS	Calls the various subroutines associated with the integration of the acoustic wave equation.
17. INTIA	Performs the integration of the acoustic wave equation using a fourth-order Runge-Kutta method.

<u>NAME</u>	<u>DESCRIPTION</u>
18. DERSUB	Subroutine which evaluates the derivative used in the Runge-Kutta integration.
19. CHSUB	Dummy subroutine called from subroutine INT1A.

TABLE II  
PRINCIPAL PROGRAM VARIABLES

<u>FORTRAN NAME</u>	<u>DEFINITION</u>
IR(1)	Number of grid points in region 1 (i.e., from Mach disc to the contact).
IR(2)	Total number of grid points (including points on both sides of both shocks and the contact except for the free stream side of the blast wave)
Z	Axial distance (in feet).
P	Natural logarithm of pressure (in psf).
U	Gas velocity (in fps).
S	Entropy (in $\text{ft}^2/\text{sec}^2 \cdot ^\circ\text{R}$ )
A	Speed of sound (in fps).
RH	Gas density (in slugs/ $\text{ft}^3$ ).
W	Velocity of surface of discontinuity.
GAM1, GAM2	Ratio of specific heats for regions 1 and 2, respectively.

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<u>FORTRAN NAME</u>	<u>DESCRIPTION</u>
CP1, CP2	Specific heat at constant pressure for regions 1 and 2, respectively.
RJET	Radius of gun bore.
XME	Muzzle exit Mach number.
PE	Muzzle exit pressure (in psf).
AE	Sound speed at muzzle exit (in fps).
PINF	Ambient pressure (in psf).
UINF	Ambient gas velocity (in fps) (usually 0.0).
AINF	Speed of sound in ambient gas (in fps).
RHINF	Ambient gas density (slugs/ft <sup>3</sup> ).
TIME	Current time (seconds).
TIMEF	Final time (seconds).
KK	Maximum number of time steps.
K0	Current number of completed time steps.
LL	Interval in the number of time steps between printing of complete flow field solution.
IPUNCH	Index for option to write final solution on TAPE12.

FORTRAN NAME DESCRIPTION

DT Time increment.

DZ Increment in axial distance.

Definition of other program variables should be self-evident from the context of their usage.

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SECTION III  
DESCRIPTION OF INPUT

**A. PUNCHED CARD FORMAT**

<u>Card Number</u>	<u>Format</u>	<u>Columns</u>	<u>Description</u>
1	15	1-5	starting step of run
	15	6-10	final step of run
	15	11-15	print interval
	15	16-20	output file creation index (0 - no output file created; 1 - output file created on TAPE12)
	15	21-25	restart indicator (0 - initial program submittal; 1 - program reads initial step from TAPE10)
	15	26-30	option for moving origin of coordinate system (0 - fixed region; 1 - moving origin)
	15	31-40	input unit index - specifies the units of the input (0 - metric; 1 - english)
	15	41-45	output unit index - determines the units of the output (0 - metric; 1 - english; 2 - non-dimensional)
	15	46-50	acoustic wave index (0 - normal program operation; 1 - only acoustic wave calculation performed)
2	E10.0	1-10	starting time of run (seconds)
	E10.0	11-20	final time of run (seconds)
	E10.0	21-30	multiplicative factor on maximum time step calculated from stability theory (recommended value is 1.0)

<u>Card Number</u>	<u>Format</u>	<u>Columns</u>	<u>Description</u>
2	E10.0	31-40	multiplicative factor on minimum grid spacing as determined by initial number of grid points (recommended value is $\leq 1.0$ )
	E10.0	41-50	multiplicative factor on minimum grid spacing to determine maximum grid spacing (recommended value is $\geq 2.0$ )
3	E10.0	1-10	radius of gun bore (m or ft)
	E10.0	11-20	coordinate system index (0 - planar; 1 - cylindrical; 2 - spherical)
	E10.0	21-30	ratio of distance from muzzle to origin of coordinate system to the distance from muzzle to the Mach disc
If the restart indicator is equal to one on card number 1, card number 4 is not required.			
4	15	1-5	last data point in region 1 (shown as IR1 in Figure 1)
	15	6-10	last data point in region 2 (shown as IR2 in Figure 1)
	E10.0	11-20	specific heat ratio in region 1
	E10.0	21-30	Specific heat ratio in region 2
	E10.0	31-40	specific heat at constant pressure in region 1 ( $\text{m}^2/\text{sec}^2 \text{ }^\circ\text{K}$ or $\text{ft}^2/\text{sec}^2 \text{ }^\circ\text{R}$ )

- — Discontinuities
- ✗ — Interior Points
- — Point where Streamline Originates

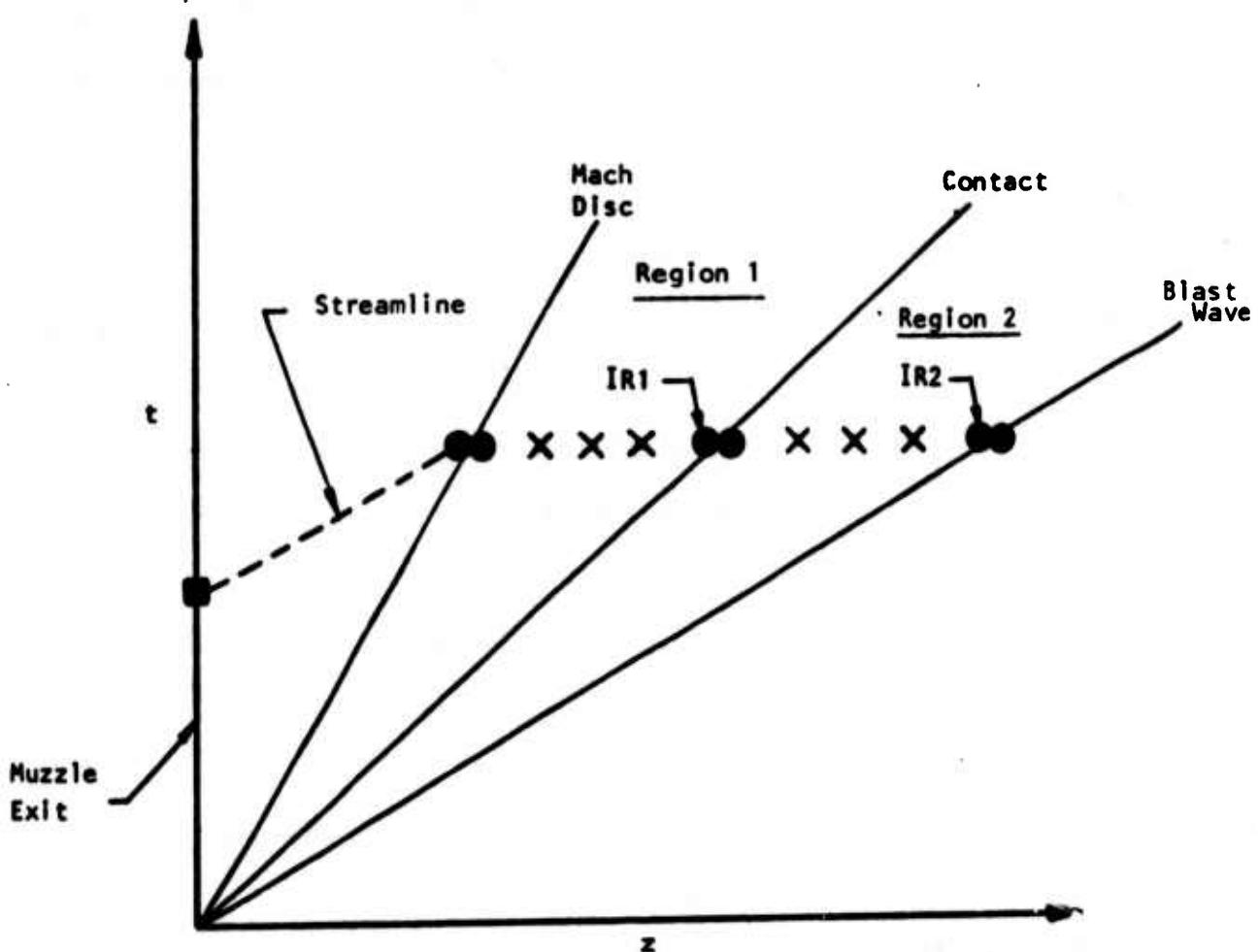


FIGURE 1. SCHEMATIC OF GRID NETWORK

<u>Card Number</u>	<u>Format</u>	<u>Columns</u>	<u>Description</u>
4	E10.0	41-50	specific heat at constant pressure in region 2 ( $\text{m}^2/\text{sec}^2 \text{ °K}$ or $\text{ft}^2/\text{sec}^2 \text{ °R}$ )
5	E10.0	1-10	ambient pressure ( $\text{N/m}^2$ or $1\text{b}/\text{ft}^2$ )
	E10.0	11-20	ambient gas velocity ( $\text{m/sec}$ or $\text{ft/sec}$ ) (typically 0.0)
	E10.0	21-30	ambient speed of sound ( $\text{m/sec}$ or $\text{ft/sec}$ )
	E10.0	31-40	blast wave Mach number at which acoustic wave analysis commences
6	15	1-5	INUM-number of points in table of muzzle exit properties (maximum of 25 points)
7	There are "INUM" of the following cards describing the muzzle exit properties.		
7a	E10.0	1-10	time (seconds)
	E10.0	11-20	pressure ( $\text{N/m}^2$ or $1\text{b}/\text{ft}^2$ )
	E10.0	21-30	speed of sound ( $\text{m/sec}$ or $\text{ft/sec}$ )
	E10.0	31-40	Mach number
8	15	1-5	INUMP-number of points in table describing the plume centerline Mach number distribution

<u>Card number</u>	<u>Format</u>	<u>Columns</u>	<u>Description</u>
--------------------	---------------	----------------	--------------------

9 The following cards describe the plume centerline Mach number distribution. A maximum of 35 points are allowed. Input the location of the point (ZZ) followed by the centerline Mach number (ZMCL) for that point. Input 4 pairs to a card and continue on the next card until all "INUMP" points have been input. A sample of the first card is shown below:

9a	E10.0	1-10	ZZ(1) - axial distance from muzzle in bore radii
	E10.0	11-20	ZMCL(1) - plume centerline Mach number at distance ZZ(1) from nozzle
	E10.0	21-30	ZZ(2)
	E10.0	31-40	ZMCL(2)
	E10.0	41-50	ZZ(3)
	E10.0	51-60	ZMCL(3)
	E10.0	61-70	ZZ(4)
	E10.0	71-80	ZMCL(4)

If the acoustic wave index is equal to zero on card 1, card number 10 is not required.

10	E10.0	1-10	the location of the contact surface (m or ft)
	E10.0	11-20	the location of the blast wave (m or ft)
	E10.0	21-30	the initial time step to be used by the Runge-Kutta integration routine (usually taken to be the final time step obtained by the stability requirements of the finite difference portion of the program) (seconds)

<u>Card Number</u>	<u>Format</u>	<u>Columns</u>	<u>Description</u>
10	E10.0	31-40	the constant in the Mach disc equation printed out at the last finite difference step
	E10.0	41-50	the constant in the blast wave equation printed out at the last finite difference step
	E10.0	51-60	the constant in the velocity curve fit printed out at the last finite difference step

B. Restart Capability - The program possesses the capability of using a solution previously stored on disk or tape to initialize the necessary variables at each grid point. To create such a file, the output file creation index (card 1) must be set equal to one (1). When the final time step is reached the necessary variables are output on TAPE12 and may be saved by the appropriate use of control cards.

To make use of this file, the restart indicator (card 1) should be set equal to one (1). With the exception of card 2 which must be changed to correspond to the time step saved on the file and card 4 which must be omitted, the program input deck is the same. The variables are then read from TAPE10 and are used for the initiation of the program. At the completion of this run, the variables can again be output on TAPE12 if desired. Care must be taken to assure that the files and tapes correspond to the desired input and output data. Typical control cards that illustrate the use of the restart capability are shown in Figure (2).

CREATION OF OUTPUT FILE

JOB CARD  
CHARGE CARD  
REQUEST(TAPE12,\*PF)  
ATTACH(LGO,DAWN,CY=1, ID=ATLXXX)  
MAP(PART)  
LGO.  
REWIND(TAPE12)  
CATALOG(TAPE12,NAMESTEPXXX,CY=1, ID=ATLXXX)  
EOR

RESTART USING OUTPUT FILE

JOB CARD  
CHARGE CARD  
REQUEST(TAPE12,\*PF)  
ATTACH(LGO,DAWN,CY=1, ID=ATLXXX)  
MAP(PART)  
ATTACH(TAPE10,NAMESTEPXXX,CY=1, ID=ATLXXX)  
REWIND(TAPE10)  
LGO.  
REWIND(TAPE12)  
CATALOG(TAPE12,NAMESTEPXXXX,CY=1, ID=ATLXXX)  
EOR

FIGURE 2. CDC SCOPE 2.1 OPERATING SYSTEM CONTROL CARDS ILLUSTRATING THE USE  
OF THE PROGRAM RESTART CAPABILITY

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SECTION IV  
DESCRIPTION OF OUTPUT

A. Output Format - The first page of the program output prints a narrative which informs the user of various input parameters such as the initial and final time steps, print interval, free stream properties etc. On the second page of output, the input values of the muzzle exit conditions and the plume centerline Mach number distribution are printed out in tabular form. The units of the input variables are specified by the input unit index as described previously in Section III.

On the following pages of the printout, starting with the initial step, the fluid dynamic properties are printed at each grid point. As a convenience to the user the locations of the Mach disc, contact surface and blast wave are specified in the output by the letters MD, C and BW respectively. At intermediate time steps where a complete output is not desired the velocity and position of the discontinuities are printed. The program has the capability of printing out the fluid variables in either metric or English units or in a non-dimensional form. The output unit index described in Section III controls which unit option is used. A table illustrating the unit option is contained in the following sub-section.

The form of the output for the acoustic wave computation is similar to that of the finite difference portion of the program except that only non-dimensional output is used. When the program reaches the time for switching to the acoustic analysis (i.e., when  $W(3)/a_\infty \leq 1.10$ ), the final finite difference time step is output along with the necessary parameters for resubmitting an acoustic wave run.

B. Identification of Variables

STEP - total number of steps taken

TIME - total elapsed time (seconds)

Description	Output Unit Index			
	0	1	2	
Z	axial distance	meter	feet	Z/D
P	pressure	N/m <sup>2</sup>	lb/ft <sup>2</sup>	P/P <sub>∞</sub>
U	velocity	m/sec	ft/sec	U/a <sub>∞</sub>
S	entropy	m <sup>2</sup> /sec <sup>2</sup> °K	ft <sup>2</sup> /sec <sup>2</sup> °R	(S-S <sub>ref</sub> )/c <sub>v</sub>
A	speed of sound	m/sec	ft/sec	a/a <sub>∞</sub>
RH	density	kg/m <sup>3</sup>	slug/ft <sup>3</sup>	ρ/ρ <sub>∞</sub>
TEMP	temperature	°K	°R	T/T <sub>∞</sub>
MACH	Mach number	-	-	-
W(1)	velocity of Mach disc	m/sec	ft/sec	W(1)/a <sub>∞</sub>
W(2)	velocity of contact	m/sec	ft/sec	W(2)/a <sub>∞</sub>
W(3)	velocity of blast wave	m/sec	ft/sec	W(3)/a <sub>∞</sub>
Z(1)	location of Mach disc	meter	feet	Z(1)/D
Z(2)	location of contact	meter	feet	Z(2)/D
Z(3)	location of blast wave	meter	feet	Z(3)/D

C. Sample Input and Output

0	10	2	0	0	0	1	2	0
0.0	0.3		E-021.0		1.0		2.0	
0.00913	2.0		0.40					
7	131.25		1.40		9507.08		6006.0	
2120.	0.0		1120.		1.10			
21								
0.876	E-031	2981E+064	2493E+031					
0.924	E-031	1776E+063	9306E+031					
0.972	E-031	0918E+063	7152E+031					
1.02	E-031	0172E+063	5246E+031					
1.068	E-039	5173E+053	3547E+031					
1.117	E-038	9409E+053	2008E+031					
1.165	E-038	4269E+053	0591E+031					
1.213	E-037	9668E+052	9272E+031					
1.261	E-037	4033E+052	6504E+031					
1.309	E-036	7934E+052	8140E+031					
1.357	E-036	2439E+052	7841E+031					
1.405	E-035	7510E+052	7592E+031					
1.453	E-035	3095E+052	7379E+031					
1.502	E-034	9135E+052	7192E+031					
1.550	E-034	5585E+052	7031E+031					
1.598	E-034	2389E+052	6890E+031					
1.646	E-033	9509E+052	6765E+031					
1.694	E-033	6913E+052	6650E+031					
1.742	E-033	4561E+052	6552E+031					
2.876	E-038	9808E+042	3228E+031					
3.876	E-033	2268E+042	0932E+031					
25								
0.0	0	0.2048	1.0371	0.4072	1.1541	0.6164	1.3203	
1.0240	1.6874	1.454	2.0468	1.8198	2.3315	2.254	2.6268	
2.916	2.9952	3.478	3.2655	4.256	3.5857	4.958	3.834	
5.786	4.1182	9.45	4.2777	7.736	4.6329	8.854	4.8369	
10.0	5.15	11.0	5.26	12.0	5.92	14.0	5.72	
16.0	6.02	16.0	6.3	20.0	6.58	40.0	8.5	
68.0	10.0							

FIGURE 3. TYPICAL INPUT IN CARD IMAGE FORMAT

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M U Z Z L E   B L A S T   A N A L Y S I S

STARTING STEP = 0

FINAL STEP = 10

PRINT INTERVAL = 10

STARTING TIME = 0. SEC.

FINAL TIME = .30000E-02 SEC.

LAST PT. REGION NO. 1 = 7

LAST PT. REGION NO. 2 = 13

MOVING COORDINATE SYSTEM OPTION = 0

RADIUS OF THE JET = .91300E-02 FT.

FLOW INDEX = 2

ORIGIN OF SPHERICAL SYSTEM = .400E+00 TIMES RJET

FREE STREAM PRESSURE = .21200E+04 LBS/FT/FT

FREE STREAM VELOCITY = 0. FT/SEC

FREE STREAM SPEED OF SOUND = .11200E+04 FT/SEC

SPECIFIC HEAT RATIO (REGION 1) = 1.25

SPECIFIC HEAT RATIO (REGION 2) = 1.40

SPECIFIC HEAT AT CONSTANT PRESSURE (REGION 1) = 9507.1 FT\*\*2/SEC\*\*2-DEG.R

SPECIFIC HEAT AT CONSTANT PRESSURE (REGION 2) = 6006.0 FT\*\*2/SEC\*\*2-DEG.R

BARREL EXIT PRESSURE = .12981E+07 LBS/FT/FT

BARREL EXIT MACH NO. = .10000E+01

BARREL EXIT SPEED OF SOUND = .42493E+04 FT/SEC

FIGURE 4. FIRST PAGE OF PROGRAM OUTPUT

## MUZZLE EXIT CONDITIONS

## PLUME CENTERLINE MACH NUMBER

TIME	PRESSURE	SOUND SP	MACH	X/R	MACH
0.	4800E-04	4249E+04	1000E+01	0.	1000E+01
1178E+07	1298E+07	1000E+01	2048E+00	1037E+01	1154E+01
3939E+04	3715E+04	1000E+01	4072E+00	1320E+01	6164E+00
1092E+07	3525E+04	1000E+01	1024E+01	1687E+01	1000E+01
1017E+07	3355E+04	1000E+01	1454E+01	2047E+01	1000E+01
9517E+06	3201E+04	1000E+01	1820E+01	2332E+01	1000E+01
8941E+06	3059E+04	1000E+01	2254E+01	2627E+01	1000E+01
8427E+06	2927E+04	1000E+01	2916E+01	2995E+01	1000E+01
7967E+06	2850E+04	1000E+01	3478E+01	3266E+01	1000E+01
7403E+06	2814E+04	1000E+01	4256E+01	3589E+01	1000E+01
6793E+06	2784E+04	1000E+01	4958E+01	3834E+01	1000E+01
6244E+06	2759E+04	1000E+01	5708E+01	4118E+01	1000E+01
5751E+06	2730E+04	1000E+01	6450E+01	4278E+01	1000E+01
5304E+06	2703E+04	1000E+01	7736E+01	4633E+01	1000E+01
4911E+06	2677E+04	1000E+01	8654E+01	4837E+01	1000E+01
4559E+06	2609E+04	1000E+01	1000E+02	5150E+01	1000E+01
4239E+06	3951E+06	1000E+01	1100E+02	5260E+01	1000E+01
7700E+05	3691E+06	1000E+01	1200E+02	5420E+01	1000E+01
8180E+05	2665E+04	1000E+01	1400E+02	5720E+01	1000E+01
8666E+05	2655E+04	1000E+01	1600E+02	6020E+01	1000E+01
2000E+02	8981E+05	1000E+01	1800E+02	6300E+01	1000E+01
3000E+02	3227E+05	1000E+01	2000E+02	6560E+01	1000E+01
			4000E+02	6500E+01	1000E+02
			6800E+02	1000E+02	

FIGURE 5. SECOND PAGE OF PROGRAM OUTPUT

TIME	TIME	TIME
-2(3)	-2(2)	-2(1)
W(3)	W(2)	W(1)
• 53250E+04	• 28578E+04	• 60717E+04
• 53192E+04	• 28863E+04	• 60654E+04
• 53134E+04	• 29149E+04	• 61163E+04
• 53079E+04	• 29433E+04	• 61465E+04
• 53025E+04	• 29717E+04	• 61760E+04
• 50097E+04	• 30000E+04	• 62049E+04
• 50212E+04	• 52973E+04	• 62331E+04
• 50324E+04	• 52922E+04	• 62614E+04
• 50435E+04	• 52874E+04	• 62889E+04
• 50538E+04	• 52826E+04	• 63151E+04
• 50641E+04	• 52781E+04	• 50641E+04
• 39092E+01	• 41932E+01	• 41932E+01
• 446991E+02	• 46791E+02	• 46791E+02
• 75120E+05	• 75587E+05	• 75587E+05
• 40001E+01	• 40250E+01	• 40250E+01
• 37263E+01	• 37728E+01	• 37728E+01
• 37030E+01	• 37958E+01	• 37958E+01
• 39751E+01	• 40741E+01	• 40741E+01
• 74650E+05	• 76513E+05	• 76513E+05
• 0	• 46084E+07	• 46084E+07
• 45380E+07	• 49321E+05	• 49321E+05
• 45029E+07	• 77874E+05	• 77874E+05
• 44668E+07	• 78321E+05	• 78321E+05
• 44308E+07	• 78764E+05	• 78764E+05

FIGURE 6. TYPICAL OUTPUT OF FINITE DIFFERENCE CALCULATION

STEP #1095		TIME = .30000E-02		
MD	Z/D	P	RMD	U
	.39490E+01			
	.18401E+03	.10000E+01	.10000E+01	0.
	.18568E+03	.10006E+01	.10011E+01	.20501E-02
	.18738E+03	.10016E+01	.10024E+01	.42083E-02
	.18912E+03	.10032E+01	.10040E+01	.64911E-02
	.19088E+03	.10053E+01	.10061E+01	.89150E-02
	.19268E+03	.10081E+01	.10085E+01	.11496E-01
	.19452E+03	.10117E+01	.10113E+01	.14252E-01
	.19638E+03	.10159E+01	.10146E+01	.17204E-01
	.19829E+03	.10207E+01	.10182E+01	.20381E-01
	.20023E+03	.10260E+01	.10220E+01	.23811E-01
	.20221E+03	.10316E+01	.10258E+01	.27524E-01
	.20423E+03	.10374E+01	.10296E+01	.31547E-01
	.20630E+03	.10432E+01	.10333E+01	.35899E-01
	.20840E+03	.10492E+01	.10369E+01	.40539E-01
	.21055E+03	.10553E+01	.10405E+01	.45406E-01
	.21274E+03	.10617E+01	.10442E+01	.50438E-01
	.21497E+03	.10684E+01	.10482E+01	.55575E-01
	.21726E+03	.10754E+01	.10525E+01	.60755E-01
	.21959E+03	.10830E+01	.10572E+01	.65970E-01
UW	.22198E+03	.10910E+01	.10623E+01	.71290E-01

END CONDITION MET

FIGURE 7. TYPICAL OUTPUT OF ACOUSTIC WAVE ANALYSIS

TM 184  
SECTION V  
PROGRAM LISTING

TM 184

PROGRAM DAWNA(INPUT,OUTPUT,PUNCH,TAPE5=INPUT,TAPE6=OUTPUT,  
1TAPE10,TAPE12)  
COMMON/A/Z(200),P(200),U(200),RH(200),S(200),A(200),W(3)  
COMMON/B/ZN(200),PN(200),UN(200),RHN(200),SN(200),AN(200),WN(3)  
COMMON/D/IR(2),GAM,XJ,CP,DZ,DT  
COMMON/F/RJET,XME,PE,AE,PINF,AINF,UINF,RHINF  
COMMON/G/KK,LL,KD,TIME,TIMEF  
COMMON/IPU/ IPUNCH  
COMMON/ZZERO/ZZERO,IMOVE,DZZERO,DTZZER,FACT  
COMMON/GA/GAM1,GAM2,CP1,CP2  
COMMON/BLAST/IBW,BWCON,SPEED,RJET2  
COMMON/DEBUG/ DEBUG  
COMMON/UNITS/IUNIT,IUDUM,IUNOUT,IUOUT,FTME,PUNIT,DEGRK,RHUNIT  
COMMON/BWANAL/HWMACH  
DATA H1/2H 1.,H2/2HMD/,H3/2H C/,H4/2HBW/  
DT=0.0  
CALL INDATA  
FACTINF=CP2\*(GAM2-1.0)/GAM2  
FACT2=FACTINF  
FACT1=CP1\*(GAM1-1.0)/GAM1  
3 CONTINUE  
IR1=IR(1)  
IR1P=IR1+1  
IR2=IR(2)  
IR2P=IR2+1  
IF(KU,GE,KK) GO TO 4  
IF((KO/LL)\*LL,NE,KO) GO TO 5  
4 CONTINUE  
WRITE(6,7) KO,TIME  
7 FORMAT(1H1,10X,\*STEP Z\*,I4,10X,\*TIME Z\*,E13.5,1X,4HSEC.,//)  
WRITE(6,8)  
8 FORMAT(4X\*I11X,\*Z\*,13X,\*P\*,13X,\*U\*,13X,\*A\*,13X,\*RHM\*,  
I11X,\*TEMP\*,10X,\*MACH\*)  
I1=IR(2)+1  
F0UM=AINF\*TIME  
DO 9 I=1,I1  
FACTREG=FACT1  
IF(I,GT,IR1) FACTREG=FACT2  
H5=H1  
IF(I,EQ,1,OR,I,EQ,2) H5=H2  
IF(I,EQ,IR1,OR,I,EQ,IR1P) H5=H3  
IF(I,GE,IR2) H5=H4  
GO TO(22,24,26),IUDOUT  
22 CONTINUE  
Z5=Z(I)\*FTME  
PS=EXP(P(I))\*PUNIT  
U5=U(I)\*FTME  
A5=A(I)\*FTME  
RS=RH(I)\*RHUNIT  
S5=S(I)\*FTME\*\*2\*DEGRK  
TEMP=(PS/(RS\*FACTREG))/FTME\*\*2/DEGRK  
GO TO 50  
24 CONTINUE  
Z5=Z(I)  
PS=EXP(P(I))  
U5=U(I)  
A5=A(I)

```

R5=RH(I)
S5=S(I)
TEMP=(P5/(R5*FACREG))
GO TO 50
26 CONTINUE
Z5=Z(I)/RJET2
P5=EXP(P(I))/PINF
U5=U(I)/AINF
A5=A(I)/AINF
RS=RH(I)/RHINF
TEMP=(P5/RS)*FACINF/FACREG
IF(I.GT.IR1) GU TO 30
S5=GAM1*(S(I)-S(1))/CP1
GO TO 40
30 CONTINUE
S5=GAM2*(S(I)-S(IR2P))/CP2
40 CONTINUE
50 CONTINUE
XMS=U(I)/A(I)
WRITE(6,10) I,H5,Z5,P5,U5,S5,A5,H5,TEMP,XMS
2000 FORMAT(8X,5E14.5)
9 CONTINUE
10 FORMAT(15,1XA2,8E14.5)
WRITE(6,12)
12 FORMAT(/)
IF(KD.GE.KK) GO TO 6
WRITE(6,21)
21 FORMAT(1)X*W(1)*10X*W(2)*10X*W(3)*10X*Z(1)*10X*Z(2)*10X*Z(3)*
110X*TIME*,11X,*DT*)
5 CONTINUE
Z1=Z(1)/RJET2
Z2=Z(IR1)/RJET2
Z3=Z(IR2)/RJET2
IF(Z3.GT.1.0)
18WCON=Z3 *(1.0-(AINF/W(3))*2)*SQRT ALOG(Z3 )
IF(Z3.GT.1.0) WCON=TIME*1.0E+06+(RJET2/AINF)*(WCON*
18SQRT ALOG(Z3))-Z3)*1.0E+06
ZMD=0.69*XME*SQRT(GAM1*PE/EXP(P(2)))
GO TO(32,34,36),IUOUT
32 CONTINUE
W1DUM=W(1)*FTME
W2DUM=W(2)*FTME
W3DUM=W(3)*FTME
Z1DUM=Z(1)*FTME
Z2DUM=Z(IR1)*FTME
Z3DUM=Z(IR2)*FTME
GO TO 38
34 CONTINUE
W1DUM=W(1)
W2DUM=W(2)
W3DUM=W(3)
Z1DUM=Z(1)
Z2DUM=Z(IR1)
Z3DUM=Z(IR2)
GO TO 38
36 CONTINUE
W1DUM=W(1)/AINF

```

```

W2DUM=W(2)/AINF
W3DUM=W(3)/AINF
Z1DUM=Z(1)/RJET2
Z2DUM=Z(IR1)/RJET2
Z3DUM=Z(IR2)/RJET2
38 CONTINUE
  WRITE(6,11) W1DUM,W2DUM,W3DUM,Z1DUM,Z2DUM,Z3DUM,TIME,DT
11 FORMAT(5X,8E14.5,2F8.2)
  IF(1BW, EQ, 0) GO TO 14
  LL=LLDUM
  ZCON=Z(IR1)
  ZBW=Z(IR2)
  RCT=Z(IR1P)-FDUM
  SGN=SIGN(1.0,RCT)
  DO 60 I=IR1P,IR2
  RCT=Z(I)-FDUM
  SGN1=SIGN(1.0,RCT)
  IF(SGN, EQ, SGN1) GO TO 59
  IRCT=1
  IRCTM=I-1
  RCTM=Z(IRCTM)-FDUM
  TERM=(Z(IRCT)/RJET2)**2*(RH(IRCT)/RHINF-1.0-U(IRCT)/SPEED)
  TERM=Z(IRCTM)/RJET2)**2*(RH(IRCTM)/RHINF-1.0-U(IRCTM)/SPEED)
  RATRCT=RCT/(RCT-RCTM)
  RHORCT=TERM+RATRCT*(TERM-TERM)
  GO TO 70
59 CONTINUE
  SG=SGN1
60 CONTINUE
  RHORCT=1.0
70 CONTINUE
  WRITE(6,1000) CONMD,BWCON,RHURCT
1000 FORMAT(//,1X,*THE CONSTANT IN THE MACH DISC EQUATION IS*,E13.5,
  //,1X,*THE CONSTANT IN THE BLAST WAVE EQUATION IS*,E13.5,
  //,1X,*THE CONSTANT IN THE VELOCITY CURVE FIT IS*,E13.5)
  CALL ACOUS(ZCON,ZBW,DT,CONMD,RHURCT)
14 CONTINUE
  CALL SETN
  CALL STEP
  CALL SHOCK(1)
  CALL SHOCK(3)
  CALL CONCT(2)
  CALL INT PT
  IF(1MOVE, EQ, 0) GO TO 20
  ZZERO=ZN(1)*FACT
  DZZERO=ZN(1)*FACT
  DTZZERO=(ZN(1)-N(1))/DT
20 CONTINUE
  I1=IK(2)+1
  DO 1 I=1,I1
  Z(I)=Z N(I)
  P(I)=P N(I)
  U(I)=U N(I)
  S(I)=S N(I)
  A(I)=A N(I)
  1  RH(I)=RH N(I)
  DO 2 I=1,3

```

```
2 W(I)=WN(I)
K0=K0+1
TIME=TIME+DT
IF((^3)/AINF),LE,WNMACH) IBW=1
IF(IBW,EQ,0) GO TO 3
CALL MUZZLE(TIME,P EXIT,A EXIT,X M EXIT)
CONMD=Z(1)/(X M EXIT*SQRT(GAM1*P EXIT/P INF))/R JET2
LLDUM=LL
LL=1
GO TO 3
6 CONTINUE
IF(IPUNCH,NE,0) CALL PUNCH
STOP
END
```

## TM 184

```

SUBROUTINE CL(IT1,BT)
COMMON/A/Z(200),P(200),US(200),RH(200),S(200),A(200),W(3)
COMMON/B/ZN(200),PN(200),UN(200),RHN(200),SN(200),AN(200),NN(3)
COMMON/D/IR(2),GAM,XJ,CP,DZ,DT
COMMON/F/RJET,XME,PE,AE,PINF,AINF,UINF,RHINF
COMMON/G/GAM1,GAM2,CP1,CP2
COMMON/G/KK,LL,KU,TIME,TIMEF
COMMON/MCL/ZZ(35),ZMCL(35),EMCL(35),INUMP
COMMON/INIT/PM(25),AM(25),XMM(25),TM(25)
COMMON/DEBUG/IDEBUG
DIMENSION ZM(1),AM(1),DXM(1),D2XM(1)
DATA MAX,ERRVAL/1,0.001/
ZE=0.0
ITER=0
RATIO=0.5
IF(BT.EQ.0.0) UN(1)=U(1)
TINIT=TIME+DT
T1=TM(1)
IF(IDEBUG.EQ.6) WRITE(6,1) IT1,ZE,RATIO,TIME,UT,TINIT,T1,BT
20 CONTINUE
TDUMS=RATIO*TINIT+(1.0-RATIO)*T1
CALL MUZZLE(TDUMS,PE,AE,XME)
UE=0.5*(XME*AE+UN(1))
TDUMP=TINIT-(ZN(1)-ZE)/UE
ERR=(TDUMS-TDUMP)/TDUMS
IF(IDEBUG.EQ.6) WRITE(6,1) ITER,TDUMS,TDUMP,PE,AE,XME,UE,ZN(1),ERR
IF(TDUMP.LT.T1.0E-10) TDUMP.GT.TINIT) GO TO 100
IF(ABS(ERR).LT.ERRVAL) GO TO 300
ITER=ITER+1
IF(ITER.GT.20) GO TO 200
RATIO=(T1-TDUMP)/(T1-TINIT)
GO TO 20
100 CONTINUE
WRITE(6,1000)
1000 FORMAT(1X,*STREAMLINE IN SUBROUTINE CL IS OUT OF BOUNDS*)
CALL EXIT
200 CONTINUE
WRITE(6,2000)
2000 FORMAT(1X,*TOO MANY ITERATIONS IN SUBROUTINE CL*)
CALL EXIT
300 CONTINUE
GAM=GAM1
I=1
ZZ1=ZN(I)/RJET+1.
G1=GAM-1.
XME2=XME*XME
F1=1.+G1*XME2/2.
G3=GAM/G1
G4=1./G1
ZM(1)=ZZ1
CALL SPLINT(ZZ,ZMCL,INUMP,ZM,MAX,XM,DXM,D2XM,EMCL)
XMCL=XM(1)
XMCL2=XMCL*XMCL
F2=1.+G1*XMCL2/2.
PPE=(F1/F2)**G3
RHRE=(F1/F2)**G4
PN(I)=PPE*PE

```

```
RHE=GAM*PE/AE/AE
RHN(I)=RHRE*RHE
AN(I)=SURT(GAM*PN(I)/RHN(I))
UN(I)=XMCL*AN(I)
SN(I)=S(1)+CP1*((ALOG(PN(I))-P(1))/GAM)-ALOG(RHN(I)/RH(I)))
PN(I)=ALOG(PN(I))
IF(IDEBUG,EJ,6) WRITE(6,1) I,PN(I),RHN(I),AN(I),UN(I),SN(I),
1S(1),P(1),RH(I)
RETURN
1 FORMAT(1I5,9E13.5)
END
```

```

SUBROUTINE CONCT(K)
COMMON/A/Z(200),P(200),U(200),RH(200),S(200),A(200),N(3)
COMMON/B/ZN(200),PN(200),UN(200),RHN(200),SN(200),AN(200),NN(3)
COMMON/C/ZI(200),PI(200),UI(200),RHI(200),SI(200),AI(200)
COMMON/D/IR(2),GAM,XJ,CP,DZ,DT
COMMON/E/BT,AL
COMMON/GA/ GAM1,GAM2,CP1,CP2
COMMON/ZZERO/ZZERO,IMOVE,DZZERO,DTZZER,FACT
IT1=1
AL=1.
BT=0.
WN(K)=W(K)
UN(I)=WN(K)
UN(I+1)=UN(I)
2 ZN(I)=Z(I)+(AL*W(K)+BT*WN(K))*DT
1 ZN(I+1)=ZN(I)
U1=WN(K)
U2=U1
GAM=GAM1
CP=CP1
CALL LPOINT(I,1.,1,1)
GAM=GAM2
CP=CP2
CALL LPOINT(I+1,-1.,-1,2)
RN=ZN(I)-ZZERO
R1=ZI(1)-ZZERO
R2=ZI(2)-ZZERO
A1=GAM1*(AL/AI(1)+BT/AN(1))
B1=GAM2*(AL/AI(2)+B1/AN(I+1))
IF(IMOVE,EW,0) GO TO 200
A4=GAM1*XJ*(AL*(UI(1)-DZZERO)/R1+BT*(UN(I)-DZZERO)/RN)
A4=A4+DT
B4=GAM2*XJ*(AL*(UI(2)-DZZERO)/R2+BT*(UN(I+1)-DZZERO)/RN)
B4=B4+DT
GO TO 210
200 A4=(AL*UI(1)/R1+BT*UN(I)/RN)*DT*XJ*GAM1
B4=(AL*UI(2)/R2+BT*UN(I+1)/RN)*XJ*GAM2*DT
210 PC1=PI(1)-(U1-UI(1))*A1-A4
PC2=PI(2)+(U2-UI(2))*B1-B4
APC1=EXP(PC1)
APC2=EXP(PC2)
ER=(APC1-APC2)/(APC1+APC2)*2.0
IF(ABS(ER),LT,1,E-04) GO TO 7
IT1=IT1+1
IF(IT1,LE,15) GO TO 777
WRITE(6,111)
111 FORMAT(5X,*TOO MANY ITERATIONS IN SUBROUTINE CONCT*)
STOP
777 IF(IT1,GT,2) GO TO 14
BET=WN(2)
ER1=ER
BET1=BET
BET1=BET1+0.01*BET
GO TO 15
14 DUM=BET1-ER1*(BET-BET1)/(ER-ER1)
ER1=ER

```

```

BET1=BET
BET=DUM
15 WN(K)=BET
ZN(I)=Z(I)+.5*(W(K)+WN(K))*DT
GO TO 1
7 PN(I)=PC1
PN(I+1)=PN(I)
UN(I)=U1
UN(I+1)=U2
SN(I)=S(I)
SN(I+1)=S(I+1)
CK=ALOG(RH(I))=P(I)/GAM1
RHN(I)=PN(I)/GAM1=(SN(I)-S(I))/CP1+CK
RHN(I)=EXP(RHN(I))
CK=ALOG(RH(I+1))=P(I+1)/GAM2
RHN(I+1)=PN(I+1)/GAM2=(SN(I+1)-S(I+1))/CP2+CK
RHN(I+1)=EXP(RHN(I+1))
AN(I)=SQRT(GAM1*EXP(PN(I))/RHN(I))
AN(I+1)=SQRT(GAM2*EXP(PN(I+1))/RHN(I+1))
IF(BT,GT,0.) GO TO 9
BT=.5
AL=.5
IT1=1
GO TO 2
9 RETURN
END

```

SUBROUTINE FS

COMMON/B/ZN(200),PN(200),UN(200),RHN(200),SN(200),AN(200),WN(3)

COMMON/D/IR(2),GAM,XJ,CP,DZ,DT

COMMON/F/RJET,XME,PE,AE,PINF,AINF,UINF,RHINF

COMMON/GA/ GAM1,GAM2,CP1,CP2

I=IR(2)+1

PN(I)=PINF

UN(I)=UINF

RHN(I)=RHINF

SN(I)=0.

AN(I)=SQRT(GAM2\*PN(I)/RHN(I))

PN(I)=ALUG(PN(I))

RETURN

END

TM 184

```

SUBROUTINE INDATA
COMMON/A/Z(200),P(200),U(200),RH(200),S(200),A(200),N(3)
COMMON/B/ZN(200),PN(200),UN(200),RHN(200),SN(200),AN(200),NN(3)
COMMON/D/IR(2),GAM,XJ,CP,DZ,DT
COMMON/F/RJET,XME,PE,AE,PINF,AINF,UINF,RHINF
COMMON/G/KK,LL,KU,TIME,TIMEF
COMMON/DZ/DZNIN,DZMAX
COMMON/DTSTAB/DTSTAB
COMMON/IPU/IPUNCH
COMMON/GA/GAM1,GAM2,CP1,CP2
COMMON/FIT/EMP(25),EMA(25),EMM(25),INUM
COMMON/INIT/PM(25),AM(25),XMM(25),TM(25)
COMMON/MCL/ZZ(35),ZMCL(35),EMCL(35),INUMP
COMMON/ZZERO/ZZERO,IMOVE,DZZERO,DTZZERO,FACT
COMMON/PRAT/PRAT
COMMON/DEBUG/IDEBUG
COMMON/BLAST/IBW,BWCON,SPEED,RJET2
COMMON/BWANAL/BW:MACH
COMMON/UNITS/IUNIT,IJDUM,IUNJUT,IUUUT,FTME,PUNIT,DEGRK,RHUNIT
DIMENSION TD(1),PD(1),DPD(1),D2PD(1)
DIMENSION H3(2),H4(2),H5(2),H6(2),H7(2)
DATA H1/4HRJET/,H2/4HZ(1)/
DATA IDEBUG/0/
DATA FACP,FACA/1.0,1.0/
DATA FTME,PUNIT,DEGRK,RHUNIT/0.3048,47.880258,1.8,515.379/
DATA HB/4HSEC./
DATA H3(1),H4(1),H5(1),H6(1),H7(1)/3H4.,9HN/M**2 ,6HM/SEC ,
10HM**2/SEC*,BH2-DEG.K /
DATA H3(2),H4(2),H5(2),H6(2),H7(2)/3HFT.,9HLBS/FT/FT,6HFT/SEC
,10HFT**2/SEC*,BH2-DEG.R/

```

```

C***** READ INPUTS *****
C***** READ(5,1)  KO,KK,LL,IFUNCH,IRSTRT,IMOVE,IUNIT,IUNOUT,IBN
C***** IUDUM=IUNIT+1
C***** IUDUT=IUNOUT+1
C***** HDUM=H1
C***** IF(IMOVE.EQ.1) HDUM=H2
C***** READ(5,2) TIME,TIMEF,DTSTAB,DZMIN,DZMAX
C***** READ(5,2) RJET,XJ,FACT
C***** IF(IUNIT.EQ.0) KJET=RJET/FTME
C***** IF(IRSTRT.GT.0) GU TO 8
C***** READ(5,3) IR(1),IR(2),GAM1,GAM2,CP1,CP2
C***** IF(IUNIT.EQ.1) GU TO 8
C***** CP1=CP1/FTME**2/DEGRK
C***** CP2=CP2/FTME**2/DEGRK
8  CONTINUE
C***** READ(5,2) PINF,UINF,AINF,BW4MACH
C***** IF(IUNIT.EQ.1) GU TO 11
C***** PINF=PINF/PUNIT
C***** UINF=UINF/FTME
C***** AINF=AINF/FTME
C***** FACP=PUNIT
C***** FACB=FTME
11  CONTINUE
C***** READ(5,1) INUM
DO 10 I=1,INUM

```



```

WRITE(6,204)
INUMQ=INUM
IF (INUMP.LT.INUMQ) INUMQ=INUMP
DO 326 I=1,INUMQ
PDUMMY=PM(I)*FACP
ADUMMY=AM(I)*FACA
WRITE(6,500) TM(I),PDUMMY,ADUMMY,XMM(I),ZZ(I),ZMCL(I)

```

```

326 CONTINUE
INUMQ=INUMQ+1
IF (INUM=INUMP) 307,308,309
307 WRITE(6,501) (ZZ(I),ZMCL(I),I=INUMQ,INUMP)
GO TO 308

```

```

309 CONTINUE
DO 327 I=INUMQ,INUM
PDUMMY=PM(I)*FACP
ADUMMY=AM(I)*FACA
WRITE(6,502) TM(I),PDUMMY,ADUMMY,XMM(I)

```

```

327 CONTINUE
308 CONTINUE
DO 12 I=1,INUMP
ZZ(I)=ZZ(I)+1.0

```

```

12 CONTINUE

```

```

C ***** C
C      INITIALIZE FLOW PROPERTIES
C *****

```

```

CALL SPLINE(ZZ,ZMCL,INUMP,EMCL)
IF (IRSTART.GT.0) GO TO 342
CALL INIT
IR1=IR(1)
IR11=IR1+1
IR2=IR(2)
IR21=IR(2)+1
IR4=IR1-1
IR5=IR1+2
IR6=IR2-1
P2=EXP(P(2))
PIR1=EXP(P(IR1))
CK=ALOG(RH(1))-P(1)/GAM1
DO 330 I=3,IR4
RAT =FLOAT(I-2)/FLOAT(IR1-2)
Z(I)=Z(2)+(Z(IR1)-Z(2))*RAT
U(I)=U(2)+(U(IR1)-U(2))*RAT
S(I)=RAT*ALOG10(S(IR1)/S(2))
S(I)=10.*S(I)
S(I)=S(2)*S(I)
P(I)=P2+(PIR1-P2)*RAT
RH(I)=ALOG(P(I))/GAM1-(S(I)-S(1))/CP1+CK
RH(I)=EXP(RH(I))
A(I)=SQRT(GAM1*P(I)/RH(I))
P(I)=ALOG(P(I))

```

```

330 CONTINUE
P2=EXP(P(IR11))
PIR2=EXP(P(IR2))
CK=ALOG(RHINF)-ALOG(PINF)/GAM2
DO 340 I=IR5,IR6
RAT =FLOAT(I-IR11)/FLOAT(IR2-IR11)
Z(I)=Z(IR11)+(Z(IR2)-Z(IR11))*RAT

```

```

U(I)=U(IR11)+(U(IR2)-U(IR11))*RAT
S(I)=RAT=ALOG10(S(IR2)/S(IR11))
S(I)=10.*S(I)
S(I)=S(IR11)*S(I)
P(I)=P2+(PIR2-P2)*RAT
RH(I)=ALOG(P(I))/GAM2-(S(I)-S(IR21))/CP2+CK
RH(I)=EXP(RH(I))
A(I)=SQRT(GAM2*P(I)/RH(I))
P(I)=ALOG(P(I))
340 CONTINUE
342 CONTINUE
RETURN
300 CONTINUE
C*****RESTART
C READ RESTART VARIABLES
C*****READ(10) IR(1),IR(2),GAM1,GAM2,CP1,CP2
READ(10) (A(N),N=1,3)
READ(10) ZZERO,DZZERO,DTZER
I1=IR(2)+1
READ(10) (Z(I),P(I),U(I),RH(I),S(I),A(I),I=1,I1)
IDUM1=IR(1)
IDUM2=IR(2)
ZDUM1=(Z(IDUM1)-Z(2))/FLOAT(IDUM1-2)
ZDUM2=(Z(IDUM2)-Z(IDUM1))/FLOAT(IDUM2-IDUM1-1)
DZMIN=DZMIN*AMIN1(ZDUM1,ZDUM2)*0.999
DZMAX=DZMAX*DZMIN*1.001
GO TO 9
1 FORMAT(14I5)
2 FORMAT(8E10.0)
3 FORMAT(2I5,4E10.0)
22 FORMAT(4E15.0)
203 FORMAT(8E12.4)
1632 FORMAT(1H1//14X*M U Z Z L E B L A S T A N A L Y S I S//)
200 FORMAT( 9X,*STARTING STEP Z*I4//9X,*FINAL STEP Z*I4//9X*PRINT
INTERVAL Z*I4//9X*STARTING TIME Z*E13.5,2X,A4,
2//9X*FINAL TIME Z*E13.5,2X,A4,//9X
3*LAST PT. REGION NO. 1 Z*I4//9X,*LAST PT. REGION NO. 2 Z*I4//9X
4 9X*MOVING COORDINATE SYSTEM OPTION Z*I5/)
201 FORMAT( 9X*RADIUS OF THE JET Z*E13.5,2X,A3,//9X
1*FLOW INDEX Z*I2//9X*ORIGIN OF SPHERICAL SYSTEM Z*E10.3,* TIMES *
1A4,/)
202 FORMAT( 9X,*FREE STREAM PRESSURE Z*E13.5,2X,A9,//9X,
1*FREE STREAM VELOCITY Z*E13.5,2X,A6, //9X,*FREE STREAM SPEED OF
1 SOUND Z*E13.5,2X,A6,/)
205 FORMAT(1H1//13X*MUZZLE EXIT CONDITIONS*19X*PLUME CENTERLINE MACH
1NUMBER*)
204 FORMAT(5X*TIME*6X*PRESSURE*4X*SOUND SP*6X*MACH*16X*X/4*9X*MACH*)
206 FORMAT(9X*SPECIFIC HEAT RATIO (REGION 1) Z*F4.2,1//9X*SPECIFIC HEAT
1 RATIO (REGION 2) Z*F4.2,1//9X,*SPECIFIC HEAT AT CONSTANT PRESSURE
2(REGION 1) Z*,F7.1,2X,A10,A8,
3 //,9X*SPECIFIC HEAT AT CONSTANT PRESSURE(REGION
42) Z*,F7.1,2X,A10,A8,/)
207 FORMAT( 9X,*BARREL EXIT PRESSURE Z*E13.5,2X,A9, //9X,*BARREL
1 EXIT MACH NO. Z*E13.5//9X*BARREL EXIT SPEED OF SOUND Z*E13.5,2X
2,A6)
500 FORMAT(4E12.4,8X,2E12.4)

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501 FORMAT(56X,2E12.4)

502 FORMAT(4E12.4)

. END

```

SUBROUTINE INT PT
COMMON/A/Z(200),P(200),U(200),RH(200),S(200),A(200),W(3)
COMMON/B/ZN(200),PN(200),UN(200),RHN(200),SN(200),AN(200),WN(3)
COMMON/C/Z1(200),PI(200),UI(200),RHI(200),SI(200),AI(200)
COMMON/D/IR(2),GAM,XJ,CP,DZ,DT
COMMON/E/BT,AL
COMMON/GA/ GAM1,GAM2,CP1,CP2
COMMON//ZZERO/ZZERO,IMOVE,DZZERO,DTZZER,FACT
AL=.5
BT=.5
K=1
GAM=GAM1
CP=CP1
NCL=2
NCM=IR(1)
88 CONTINUE
NCL1=NCL+1
NCM1=NCM-1
M=NCM+1
DO 20 I=NCL,NCM
U I(I)=U (I)
P I(I)=P (I)
S I(I)=S (I)
A I(I)=A (I)
20 RHI(I)=RH(I)
DX=1./FLOAT(NCM-NCL)
LOOP=0
DEL=ZN(NCM)-ZN(NCL)
DD=1./DEL
8 XI=0,
IF(LOOP .EQ.1) GO TO 30
WNK=WN(K)
WNKP=WN(K+1)
GO TO 31
30 WNK=WN(K)
WNKP=WN(K+1)
31 CONTINUE
DO 1 I=NCL1,NCM1
XI=XI+DX
NM1=I-LOOP
NP1=NM1+1
EE=DD*EXP(PI(I))/RHI(I)
CC=DD*(UI(I)+(XI-1.)*WNK -XI*WNKP )
FF=DD*GAM
PX=(PI(NP1)-PI(NM1))/DX
UX=(UI(NP1)-UI(NM1))/DX
SX=(SI(NP1)-SI(NM1))/DX
R1=Z(I)
IF(LOOP.GT.0) R1=ZN(I)
R1=R1-ZZERO
URELI=UI(I)-FLOAT(IMOVE)*DZZERO
PT=-(CC*PX+FF*UX+XJ*GAM*URELI/R1)
UT=-(CC*UX+EE*PX)
ST=CC*SX
ZN(I)=ZN(NCL)+XI*DEL
IF(LOOP.EQ.1) GO TO 7
PN(I)=PI(I)+PT*DT

```

```

C UN(I)=UI(I)+UT*DT
SN(I)=S(I)+SI*DT
IO= 1
IF(UN(I).LT.0.) IO=-1
CALL LPOINT(I,0.,IU,M)
SN(I)=SI(M)
CK=ALUG(RH(I))-P(I)/GAM
RHN(I)=PN(I)/GAM-(SN(I)-S(I))/CP+CK
RHN(I)=EXP(RHN(I))
AN(I)=SQRT(GAM*EXP(PN(I)))/RHN(I)
GO TO 2
7 PN(I)=.5*(PI(I)+P(I)+PT*DT)
UN(I)=.5*(UI(I)+U(I)+UT*DT)
C SN(I)=.5*(SI(I)+S(I)+SI*DT)
IO= 1
IF(UN(I).LT.0.) IO=-1
CALL LPOINT(I,0.,IU,M)
SN(I)=SI(M)
CK=ALUG(RH(I))-P(I)/GAM
RHN(I)=PN(I)/GAM-(SN(I)-S(I))/CP+CK
RHN(I)=EXP(RHN(I))
AN(I)=SQRT(GAM*EXP(PN(I))/RHN(I))
2 CONTINUE
1 CONTINUE
IF(LOOP.EQ.1) GO TO 10
DO 3 I=NCL,NCM
U I(I)=U N(I)
P I(I)=P N(I)
S I(I)=S N(I)
A I(I)=A N(I)
RHI(I)=RHN(I)
3 CONTINUE
LOOP=1
GO TO 8
10 IF(K.EQ.2) GO TO 66
K=K+1
NCL=NCM+1
NCM=IH(2)
GAM=GAM2
CP=CP2
GO TO 88
66 RETURN
END

```

```

SUBROUTINE LPOINT(N,OPT,IO,M)
COMMON/A/Z(200),P(200),U(200),RH(200),S(200),A(200),W(3)
COMMON/B/ZN(200),PN(200),UN(200),RHN(200),SN(200),AN(200),NN(3)
COMMON/C/ZI(200),PI(200),UI(200),RHI(200),SI(200),AI(200)
COMMON/D/IR(2),GAM,XJ,CP,DZ,DT
COMMON/E/HT,AL
KJ=1
IF(OPT.EQ.0.) KJ=0
FIO=FLOAT(IO)
I=N
4 K=I-IO
E1=HT*(UN(I)+OPT*AN(I))
EM1R=AL*(U(I)+OPT*A(I))+E1
EM1L=AL*(U(K)+OPT*A(K))+E1
ZB=(Z(I)+Z(K))/2.
KIP=0
88 CONTINUE
RAT=(ZH-Z(I))/(Z(K)-Z(I))
IF(ABS(RAT).LE.1.) GO TO 1
WRITE(6,111)
111 FORMAT(5X,*CHARACTERISTIC SHOT BACK IS OUT OF RANGE IN SUBROUTINE
LPOINT*)
WRITE(6,113) I,K,IO,ZB,Z(K),Z(I)
113 FORMAT(* I,K,IO,ZB,Z(K),Z(I)=315,3E15.4)
1 EM1=EM1H+RAT*(EM1L-EM1H)
ZBT=ZH
ZB=ZN(I)-EM1*DT
IF(KJ.GT.0) GO TO 2
IF(FIU*Z(I)-FIU*ZB) 3,2,2
3 KJ=1
IO=-IO
GO TO 4
2 CONTINUE
TESTZ=ABS((ZH-ZBT)/(Z(I)-Z(K)))
IF(TESTZ.LT..01) GO TO 86
KIP=KIP+1
IF(KIP.LE.15) GO TO 88
WRITE(6,112)
112 FORMAT(5X,*100 MANY ITERATIONS IN SUBROUTINE LPOINT*)
STOP
86 CONTINUE
ZI(N)=ZB
U I(M)=U (I)+RAT*(U (K)-U (I))
SI(M)=S(I)+RAT*(S(K)-S(I))
P I(M)=P (I)+RAT*(P (K)-P (I))
CK=ALOG(RH(I ))-P(I )/GAM
RHI(M)=PI(M)/GAM-(SI(M)-S(I))/CP+CK
RHI(M)=EXP(RHI(M))
AI(M)=SQRT(GAM*EXP(PI(M))/RHI(M))
RETURN
END

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```
SUBROUTINE MUZZLE(TIME,PE,AE,XME)
COMMON/INIT/PM(25),AM(25),XMM(25),TM(25)
COMMON/FIT/EVP(25),EMA(25),EMM(25),INUM
DIMENSION TD(1),PD(1),DPD(1),D2PD(1)
MAX=1
TD(1)=TIME
CALL SPLINT(TM,PM,INUM,TD,MAX,PD,DPD,D2PD,EMP)
PE=PD(1)
CALL SPLINT(TM,AM,INUM,TD,MAX,PD,DPD,D2PD,EMA)
AE=PD(1)
CALL SPLINT(TM,XMM,INUM,TD,MAX,PD,DPD,D2PD,EMM)
XME=PD(1)
RETURN
END
```

```
SUBROUTINE PUNCH
COMMON/A/Z(200),P(200),U(200),RH(200),S(200),A(200),W(3)
COMMON/D/IR(2),GAM,XJ,CP,DZ,DT
COMMON/GA/ GAM1,GAM2,CP1,CP2
COMMON/ZZERO/ZZERO,IMOVE,DZZERO,DTZZER,FACT
WRITE(12) IR(1),IR(2),GAM1,GAM2,CP1,CP2
WRITE(12) (W(N),N=1,3)
WRITE(12) ZZERO,DZZERO,DTZZER
I1=IR(2)+1
WRITE(12) (Z(I),P(I),U(I),RH(I),S(I),A(I),I=1,I1)
RETURN
END
```

SUBROUTINE SETN

COMMON/A/Z(200),P(200),U(200),RH(200),S(200),A(200),W(3)  
COMMON/B/ZN(200),PN(200),UN(200),RHN(200),SN(200),AN(200),WN(3)  
COMMON/D/IR(2),GAM,XJ,CP,DZ,DT

I1=IR(2)+1

DO 1 I=1,I1

A N(I)=A (I)

P N(I)=P (I)

U N(I)=U (I)

S N(I)=S (I)

RHN(I)=RH(I)

1 Z N(I)=Z (I)

DO 2 I=1,3

2 WN(I)=W(I)

RETURN

END

```

SUBROUTINE SHOCK(K)
COMMON/A/Z(200),P(200),U(200),RM(200),S(200),A(200),W(3)
COMMON/B/ZN(200),PN(200),UN(200),RHN(200),SN(200),AN(200),WN(3)
COMMON/C/ZI(200),PI(200),UI(200),RHI(200),SI(200),AI(200)
COMMON/D/IR(2),GAM,XJ,CP,DZ,DT
COMMON/E/BT,AL
COMMON/GA/ GAM1,GAM2,CP1,CP2
COMMON/ZZERO/ZZERO,MOVE,UZZERO,DZZER,FACT
COMMON/DEBUG/IDEBUG
COMMON/GLASI/IBW,B,CUN,SPEED,RJET2
REAL MREL,MREL1,MREL2
IT=1
M#1
IF(K,EQ,3) GO TO 3
GAM=GAM1
CP=CP1
I#1
N#2
OPT#=1.0
GO TO 4
3 CONTINUE
GAM=GAM2
CP=CP2
I=IR(2)+1
N=IR(2)
OPT=1.0
4 CONTINUE
G2=GAM+1.
G1=GAM-1.
AL#1.
BT#0.
WN(K)=W(K)
MREL=(U(I)-W(K))/A(I)
IO=OPT
LEN=IFIX(OPT)
IFLAG#0
STEP=0.12
IFLGM#0
2 CONTINUE
IF(ABS(MREL),LE,1.0,AND,IT,GT,2) IFLGM=1
IF(IFLGM,EQ,1) MREL=SIGN(1.0,MREL)
ZN(I)=Z(I)+(AL*N(K)+BT*WN(K))/DT
1 ZN(N)=ZN(I)
IF(K,EQ,3) GO TO 6
CALL CL(IT1,BT)
WN(K)=UN(I)-MREL*AN(I)
GO TO 5
6 CONTINUE
CALL FS
IF(IBW,EQ,0) WN(K)=UN(I)-MREL*AN(I)
ZDUM=ZN(N)/RJET2
IF(IBW,EQ,1) WN(K)=SPEED/SQRT(1.0-BWCON*(SQRT(ALOG(ZDUM ))/ZDUM ))
IF(IBW,EQ,1,AND,BT,EQ,0.0) GO TO 6
5 CONTINUE
V1=UN(I)-WN(K)
RH1=RHN(I)
P1=PN(I)

```

```

XM1=V1/AN(I)
XM12=XM1*XM1
V12=G2*XM12/(G1*XM12+2.)
V2=V1/V12
RH2=RH1*V12
P2=ALOG((2.*GAM*XM12-G1)/G2)+P1
U2=V2+WN(K)
AP2=EXP(P2)
IF(IFLG.M, EQ, 0) GO TO 7
IF(JBN, EQ, 1, AND, K, EG, 3) GO TO 7
CALL LPOINT(N, UPT, ID, M)
A6=GAM/(AL*AI(I)+HT*AN(N))
IF(IMOVE, EQ, 0) GO TO 200
A7=GAM*XJ*(AL*(U1(I)-DZERO)/(Z1(I)-ZZERO))
I+BT*(UN(N)-DZERO)/(ZN(N)-ZZERO)
A7=A7*DT
GO TO 210
200 A7=XJ*GAM*DT*(AL*U1(I)/(Z1(I)-ZZERO)+BT*UN(N)/(ZN(N)-Z
ZERO))
210 P2S=PI(1)-OPT*A6*(U2-U1(I))
P2S=P2S-A7
AP2S=EXP(P2S)
ERROR=(AP2-AP2S)/(AP2+AP2S)*2.0
IF(IDEBUG, EQ, 2) WRITE(6,10) IT, I, N
IF(IDEBUG, EQ, 2) WRITE(6,20) MREL, ZN(I), ZN(N), WN(K), UN(I), AN(I),
V1, RH1, P1, XM1
IF(IDEBUG, EQ, 2) WRITE(6,20) XM12, V12, V2, RH2, P2, V2, AP2, A6, A7, P2S
IF(IDEBUG, EQ, 2) WRITE(6,20) AP2S, ERROR, XJ
IF(ABS(ERROR), LT, 1.E-04) GO TO 7
S1=SIGN(1., ERROR)
SSTEP=SIGN(1., STEP)
IF(ABS(MREL), LE, 1.05, AND, ABS(STEP), NE, 0.01) STEP=0.0025*SSTEP
IT=IT+1
IF(IT, GT, 75) GO TO 110
IF(IT, GT, 2) GO TO 40
50 CONTINUE
MREL1=MREL
ER1=ERROR
S2=S1
MREL=MREL+STEP
GO TO 2
40 CONTINUE
IF(S1, NE, S2, OR, IFLAG, EQ, 1) GO TO 45
IF(ABS(ERROR), LE, ABS(ER1)) GO TO 50
STEP=-STEP
GO TO 50
110 CONTINUE
WRITE(6,111) K, MREL, U2, RH2, P2
111 FORMAT(5X, *TOO MANY ITERATIONS IN SUBROUTINE SMUCK*, 2X, *K=*, 15, 2X,
1*MREL=*, E13.5, 2X, *U2=*, E13.5, 2X, *RH2=*, E13.5, 2X, *P2=*, E13.5)
STOP
120 CONTINUE
WRITE(6,121)
121 FORMAT(5X, *ABSOLUTE VALUE OF MACH NUMBER IS LESS THAN ONE*)
CALL EXIT
45 CONTINUE
MREL2=MREL1-ER1*(MREL-MREL1)/(ERROR-ER1)

```

```
MREL1=MREL
ER1=ERROR
MREL=MREL2
IFLAG=1
GO TO 2
7 UN(N)=U2
RHN(N)=RH2
PN(N)=P2
SN(N)=SN(I)+CP*((PN(N)-PN(I))/GAM- ALOG(RHN(N)/RHN(I)))
AN(N)=SQRT(GAM*AP2/RH2)
IF(BT.GT.0.) GO TO 9
8 CONTINUE
BT=.5
AL=.5
IT=1
IFLAG=0
STEP=.02
IFLG=0
GO TO 2
9 RETURN
10 FORMAT(14I5)
20 FORMAT(10E13.5)
END
```

```

SUBROUTINE SPLINE(X,Y,N,EM)
DIMENSION X(N),Y(N)
DIMENSION SH(25),G(25),EM(25)
DATA SB(1),G(1)/-0.5,0.0/
N0=N-1
IF(N0.LT.2) GO TO 20
DO 10 I=2,NU
A = (X(I)-X(I-1))/6.
C = (X(I+1)-X(I))/6.
W = 2.*(A+C)-4.*SH(I-1)
SB(I) = C/W
F = (Y(I+1)-Y(I))/(X(I+1)-X(I))-(Y(I)-Y(I-1))/(X(I)-X(I-1))
10 G(I) = (F-A*G(I-1))/W
20 EM(N) = G(N-1)/(2.+SB(N-1))
DO 30 I=2,N
K = N+1-I
30 EM(K) = G(K)-SB(K)*EM(K+1)
RETURN
END

```

```

SUBROUTINE SPLINT(X,Y,N,Z,MAX,YINT,DYDX,DY2DX,EM)
DIMENSION X(N),Y(N),Z(MAX),YINT(MAX),DYDX(MAX),DY2DX(MAX)
DIMENSION EM(25)
DATA SRW/0/
INTEGER SRW
III = SRW
DO 140 I=1,MAX
K=2
IF(Z(I)=X(1)) 70,60,90
60  YINT(I)=Y(1)
SK = X(K)-X(K-1)
GO TO 130
70  IF(Z(I).GE.(1.1*X(1)-.1*X(2))) GO TO 120
C   WRITE (6,1000) Z(I)
SRW = 16
GO TO 120
80  K=N
IF(Z(I).LE.(1.1*X(N)-.1*X(N-1))) GO TO 120
C   WRITE (6,1000) Z(I)
SRW = 16
GO TO 120
90  IF(Z(I)=X(K)) 120,100,110
100 YINT(I)=Y(K)
SK = X(K)-X(K-1)
GO TO 130
110 K=K+1
IF(K=N) 90,90,80
120 CONTINUE
SK = X(K)-X(K-1)
YINT(I) = EM(K-1)*(X(K)-Z(I))*3/6./SK + EM(K)*(Z(I)-X(K-1))*3/6.
+ /SK+(Y(K)/SK - EM(K)*SK/6.)*(Z(I)-X(K-1))+(Y(K-1)/SK - EM(K-1)
+ *SK/6.)*(X(K)-Z(I))
130 DYDX(I)=-EM(K-1)*(X(K)-Z(I))*2/2.0/SK + EM(K)*(X(K-1)-Z(I))*2/2.
+ /SK+(Y(K)-Y(K-1))/SK -(EM(K)-EM(K-1))*SK/6.
DY2DX(I)=EM(K-1)*(X(K)-Z(I))/SK+EM(K)*(Z(I)-X(K-1))/SK
140 CONTINUE
MXA = MAX0(N,MAX)
C   IF(SRW.EQ.16) WRITE(6,1010) N,MAX,(X(I),Y(I),Z(I),YINT(I),DYDX(I),
C   +DY2DX(I),I=1,MAX)
SRW = III
RETURN
1000 FORMAT (54H SPLINT USED FOR EXTRAPOLATION. EXTRAPOLATED VALUE = ,
+G14.6)
1010 FORMAT (2X,21HNU. OF POINTS GIVEN =,I3,50H, NO. OF INTERPOLATED PO
+INTS =,I3/10X,1HX,19X,1HY,16X,11HX=INTERPOL.,9X,11HY=INTERPOL.,
+8X,14HDYDX=INTERPOL.,6X,15HDY2DX=INTERPOL./,(6E20,8))
END

```

```

SUBROUTINE STEP
COMMON/A/Z(200),P(200),U(200),RH(200),S(200),A(200),W(3)
COMMON/D/IR(2),GAM,XJ,CP,DZ,DT
COMMON/G/KK,LL,KU,TIME,TIMEF
COMMON/DZ/ DZMIN,DZMAX
COMMON/GA/ GAM1,GAM2,CP1,CP2
COMMON/DTSTAB/DTSTAB
DT =1.E+06
K=1
NCL=2
NCM=IR(1)
5 CONTINUE
DZ=(Z(NCM)-Z(NCL))/FLOAT(NCM-NCL)
IF(DZ.GT.DZMIN) GO TO 10
IF(NCL.NE.2) GO TO 20
IF(IR(1).LE.4) GO TO 10
GAM=GAM1
CP=CP1
CALL INTER(1,IR(1),-1)
IR(1)=IR(1)+1
NCM=IR(1)
GO TO 5
20 GAM=GAM2
CP=CP2
IF(IR(2)-IR(1).LE.3) GO TO 10
CALL INTER(IR(1),IR(2),-1)
NCM=IR(2)
GO TO 5
10 CONTINUE
IF(DZ.LT.DZMAX) GO TO 40
IF(IR(2).GE.199) GO TO 40
IF(NCL.NE.2) GO TO 30
GAM=GAM1
CP=CP1
CALL INTER(1,IR(1),+1)
IR(1)=IR(1)+1
NCM=IR(1)
GO TO 5
30 GAM=GAM2
CP=CP2
CALL INTER(IR(1),IR(2),+1)
NCM=IR(2)
GO TO 5
40 CONTINUE
DO 7 I=NCL ,NCM
U1= U(I)
A1= A(I)
X1=AH5(U1/A1)+1
DT1=DZ/A1/X1
IF(DT1.LT.DT) DT=DT1
7 CONTINUE
IF(K.EQ.2) GO TO 8
IF(IR(2)-IR(1).LE.2) GO TO 8
K=K+1
NCL=NCM+1
NCM=IR(2)
GO TO 5

```

8 CONTINUE  
DT=DT+DTSTAB  
TT=TIME+DT  
IF(TT,LT,TIMEF) RETURN  
DT=TIMEF-TIME  
KK=KO  
RETURN  
END

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```

SUBROUTINE INTER(II,IF,IOPT)
COMMON/A/Z(200),P(200),U(200),RH(200),S(200),A(200),W(3)
COMMON/B/ZN(200),PM(200),UN(200),RHN(200),SN(200),AN(200),NN(3)
COMMON/D/IR(2),GAM,XJ,CP,DZ,DT
COMMON/F/RJET,XME,PE,AE,PINF,AINF,UINF,RHINF
II1=II+1
II2=II+2
IFN=IF+IOPT
JI=II1
IDUM=1
IF(IF.EQ.IR(2)) IDUM=IR(2)+1
DO 1 I=II2,IFN
  RATE=FLOAT(I-II1)/FLOAT(IFN-II1)
  ZN(I)=Z(JI)+((Z(IF)-Z(II1))*RAT)
  DO 10 J=JI,IF
    JU=J
    IF(Z(J)-ZN(I)) 10,3,2
  10 CONTINUE
  IF(Z(IF)+.0001.GT.ZN(I)) GO TO 3
  WRITE(6,100)
100 FORMAT(* ERROR IN SUBROUTINE INTER*)
  STOP
3  UN(I)=U(JU)
  SN(I)=S(JU)
  PN(I)=P(JU)
  AN(I)=A(JU)
  RHN(I)=RH(JU)
  JI=JU
  GO TO 1
2  JL=JU-1
  RAT=(ZN(I)-Z(JL))/(Z(JU)-Z(JL))
  UN(I)=U(JL)+((U(JU)-U(JL))*RAT)
  RHN(I)=RH(JL)+((RH(JU)-RH(JL))*RAT)
  P2=EXP(P(JL))
  PN(I)=P2+((EXP(P(JU))-P2)*RAT)
  SN(I)=S(IDUM)+CP*(( ALOG(PN(I))-P(IDUM))/GAM=ALOG(RHN(I)/RH(IDUM)))
  AN(I)=SQRT(GAM*PN(I)/RHN(I))
  PN(I)=ALOG(PN(I))
  JI=JU
1  CONTINUE
  ,51=IF+1
  IR21=IR(2)+1
  DO 4 I=IF1,IR21
    J=I+IOPT
    ZN(J)=Z(I)
    UN(J)=U(I)
    SN(J)=S(I)
    PN(J)=P(I)
    AN(J)=A(I)
4  RHN(J)=RH(I)
  IR(2)=IR(2)+1
  IR21=IR(2)+1
  DO 5 I=II2,IR21
    Z(I)=ZN(I)
    U(I)=UN(I)
    S(I)=SN(I)
    P(I)=PN(I)

```

A (I)=A N(I)  
S RH(I)=RHN(I)  
RETURN  
END

```

SUBROUTINE INIT
COMMON/A/Z(200),P(200),U(200),RH(200),S(200),A(200),W(3)
COMMON/B/ZN(200),PN(200),UN(200),RHN(200),SN(200),AN(200),NN(3)
COMMON/D/IR(2),GAM,XJ,CP,DZ,DT
COMMON/F/RJET,XME,PE,AE,PINF,AINF,UINF,RHINF
COMMON/G/ GAM1,GAM2,CP1,CP2
COMMON/DZ/DZMIN,DZMAX
COMMON/FIT/EMP(25),EMA(25),EMM(25),INUM
COMMON/INIT/PM(25),AM(25),XMM(25),TM(25)
COMMON/MCL/ZZ(35),ZNCL(35),EMCL(35),INUMD
COMMON/G/KK,LL,KI,TIME,TIMEF
COMMON/ZZERO/ZZERO,IMOVE,DZZERO,DTZZER,FACT
COMMON/DEBUG/IDEBUG
DIMENSION ZM(1),XM(1),DXM(1),D2XM(1)
DATA TSTEP,ERRVAL,MAX/1.E+03,0.001,1/
DATA TJUMP/1,2/
C*****SUBROUTINE INIT INITIALIZES ALL POINTS UPSTREAM AND DOWNSTREAM OF
C THE DISCONTINUITIES
C*****
DMIN=RJET
JDUM=INT(XJ+1.5)
JM=JDUM-1
G1=GAM1-1.0
G3=GAM1/G1
G4=1.0/G1
G5=GAM1+1.0
G6=GAM2-1.0
G7=GAM2+1.0
IR1=IR(1)
IH1P=IR1+1
IH2=IH(2)
IH2P=IR2+1
ZE=ZZ(1)*RJET-RJET
TINIT=TM(1)+(TM(INUM)-TM(1))/TSTEP
ITERT=1
IF(IMOVE,EQ,1) GO TO 4
ZZERO=FACT*RJET
DZZERO=0.0
DTZZER=0.0
4 CONTINUE
IF(IMOVE,EQ,1,AND,FACT,EQ,0.0,0,AND,JDUM,NE,1) TJUMP=3,0
5 CONTINUE
T1=TM(1)
CALL MUZZLE(T1,P1,A1,XM1)
U1=XM1*A1
T2=TINIT
CALL MUZZLE(T2,P2,A2,XM2)
U2=XM2*A2
DELT=TINIT-TM(1)
ITERZ=1
ZLOW=ZE
ZUP=ZE+ U1*TINIT
10 CONTINUE
ZMD=ZLOW+0.2*(ZUP-ZLOW)
C*****DETERMINE MACH NUMBER AT MACH DISC-POINT 1

```

```

***** ZM(1)=ZMD/RJET+1.0
***** CALL SPLINT(ZZ,ZMCL,INUMP,ZM,MAX,XM,DXM,D2XM,EMCL)
***** XMMD=XM(1)
***** C LOCATE ORIGIN OF STREAMLINE AT MUZZLE EXIT
***** C***** ITERS=1
***** RATIO=0.5
20 CONTINUE
TDUMS=RATIO*TINIT+(1.0-RATIO)*T1
UE=RATIO*U2+(1.0-RATIO)*U1
TDUMP=TINIT-(ZMD-ZE)/UE
ERR=(TDUMS-TDUMP)/TDUMS
IF(TDUMP.LT.T1.OR.TDUMP.GT.TINIT) GO TO 650
IF(ABS(ERR).LT.ERRVAL) GO TO 300
ITERS=ITERS+1
IF(ITERS.GT.20) GO TO 200
RATIO=(T1-TDUMP)/(T1-TINIT)
GO TO 20
200 CONTINUE
IF(IDEBUG.EQ.3) WRITE(6,2000)
2000 FORMAT(1X,*TOO MANY ITERATIONS FOR LOCATION OF STREAMLINE IN SUBROUTINE INIT*)
CALL EXIT
300 CONTINUE
PE=RATIO*P2+(1.-RATIO)*P1
AE=RATIO*A2+(1.-RATIO)*A1
XME=RATIO*XMD+(1.-RATIO)*XM1
***** C DETERMINE PROPERTIES AT POINT 1
***** F1=1.0+0.5*G1*XME**2
***** F2=1.0+0.5*G1*XMD**2
***** PPE=(F1/F2)**G3
***** RHRE=(F1/F2)**G4
***** P(1)=PPE*PE
***** RHE=GAM1*PE/AE/AE
***** RH(1)=RHRE*RHE
***** A(1)=SQRT(GAM1*P(1)/RH(1))
***** U(1)=XMD*A(1)
***** WITER=U(1)-1.10*A(1)
***** IF(1MOVE.EQ.0) GO TO 15
***** ZZERO=ZMD*FACT
***** DZZERO=WITER*FACT
***** D1ZZER=0.0
15 CONTINUE
IF(IDEBUG.EQ.3)
1 WRITE(6,1) PE,AE,XME,P(1),RH(1),A(1),U(1),WITER
***** C DETERMINE PROPERTIES AT POINT 2
***** XMPT1=1.10
***** XI=(2.*GAM1*XMP1**2-G1)/G5
***** P(2)=XI*P(1)
***** FXI=(G5*XI+G1)/(G1*XI+G5)
***** U(2)=WITER+(U(1)-WITER)/FXI

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RH(2)=FXI★RH(1)
CONSMOD=RH(2)★U(2)★(ZMD-ZZERO)★JM
IF(IDEBUG.EQ.3)
 1 WRITE(6,1) XMPT1,X1,P(2),FXI,U(2),RH(2)
C*****DETERMINE BLAST WAVE PROPERTIES
C*****XIBW=P(2)/PINF
FXI1=(G7★XIBW+G6)/(2.★GAM2)
VSBW=SQRT(FXI1★GAM2★PINF/RHINF)
FXI2=(G7★XIBW+G6)/(G6★XIBW+G7)
RH(IR2)=FXI2★RHINF
UDUM2=VSBW-VSBW/FXI2
ZSHK=ZE+VSBW★DELT
CONSBW=RH(IR2)★UDUM2★(ZSHK-ZZERO)★JM
C*****DETERMINE PROPERTIES AT MD SIDE OF CONTACT
C*****CONST=CONSMOD★DELT/RH(2)
GO TO (102,101,100),JDUM
100 CONTINUE
ZC=0.5★(ZMD+ZSHK)
ITERZC=1
30 CONTINUE
YDUM=ZC★3-2.★ZZERO★ZC★2+ZZERO★2★ZC-CONST
DYDUM=3.0★ZC★2-4.0★ZZERO★ZC+ZZERO★2
ZCDUM=ZC-(YDUM/DYDUM)
ERRZC=(ZC-ZCDUM)/ZC
IF(IDEBUG.EQ.5) WRITE(6,3) ITERZC,ZC,ZCDUM,YDUM,DYDUM,ERRZC
IF(ITERZC.GT.20) GO TO 40
ITERZC=ITERZC+1
IF(ABS(ERRZC).LE.ERRVAL) GO TO 50
ZC=ZCDUM
GO TO 30
40 CONTINUE
IF(IDEBUG.EQ.3) WRITE(6,5000)
5000 FORMAT(1X,*TOO MANY ITERATIONS FOR CONTACT POSITION IN SUBROUTINE
1INIT)
GO TO 650
101 CONTINUE
ZC=(ZZERO+SQRT(ZZERO★2+4.*CONST))/2.0
GO TO 50
102 CONTINUE
ZC=CONST
50 CONTINUE
UC1=ZC/DELT
PC1=P(2)
RHC1=RH(2)
C*****DETERMINE PROPERTIES AT BW SIDE OF CONTACT
C*****PC2=PC1
UC2=UC1
RHC2=CONSBW/(UC2*(ZC-ZZERO)★JM)
C*****CHECK SOLUTION USING DENSITY AT BW SIDE OF CONTACT

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ITERZ=ITERZ+1
ERROR=(RH(IR2)-RHC2)/RH(IR2)
IF(IDEBUG.EQ.3)
1 WRITE(6,3) ITERZ,XIBW,VSBW,RH(IR2),RHC2,ERROR,ZMD,ZC,ZSHK
IF(ABS(ERROR).LE.ERRVAL) GO TO 700
IF(ITERZ.GT.20) GO TO 600
IF(ERROR.GT.0.0) GO TO 500
ZLOW=ZMD
GO TO 10
500 CONTINUE
ZUP=ZMD
GO TO 10
600 CONTINUE
IF(IDEBUG.EQ.3) WRITE(6,3000)
3000 FORMAT(1X,*TOO MANY ITERATIONS FOR SHOCK VELOCITY IN SUBROUTINE IN
1 IT*)
650 CONTINUE
ITERT=ITERT+1
IF(ITERT.GT.20) GO TO 675
TINIT=TJUMP+TINIT
GO TO 5
675 CONTINUE
WRITE(6,4000)
4000 FORMAT(1X,*TOO MANY ITERATIONS IN SUBROUTINE INIT FOR THE INITIAL
1 TIME STEP*)
CALL EXIT
700 CONTINUE
IF(ZC.LT.ZMD.OR.ZC.GT.ZSHK) GO TO 600
IF(ZMD.GE.DMIN) GO TO 750
TINIT=TJUMP+TINIT
GO TO 5
750 CONTINUE
C*****DETERMINE ALL FLOW PROPERTIES AT THE DISCONTINUITIES*****
C*****TIME=TINIT
W(1)=WITER
Z(1)=ZMD
S(1)=0.0
A(1)=SQRT(GAM1*P(1)/RH(1))
P(1)= ALOG(P(1))
Z(2)=Z(1)
S(2)=S(1)+CP1*((ALOG(P(2))-P(1))/GAM1-ALOG(RH(2)/RH(1)))
A(2)=SQRT(GAM1*P(2)/RH(2))
P(2)= ALOG(P(2))
Z(IR1)=ZC
S(IR1)=S(2)
P(IR1)= ALOG(PC1)
RH(IR1)=RHC1
U(IR1)=UC1
A(IR1)=SQRT(GAM1*PC1/RHC1)
W(2)=UC1
Z(IR2P)=ZSHK
P(IR2P)= ALOG(PINF)
U(IR2P)=UINF
RH(IR2P)=RHINF
S(IR2P)=0.0

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A(IR2P)=SQRT(GAM2*PINF/RMINF)
W(3)=VSBW
Z(IR2)=Z(IR2P)
P(IR2)=P(2)
U(IR2)=UDUM2
S(IR2)=CP2*((P(IR2)-P(IR2P))/GAM2-ALOG(RH(IR2)/RMINF))+S(IR2P)
A(IR2)=SQRT(GAM2*EXP(P(IR2))/RH(IR2))
Z(IR1P)=Z(IR1)
P(IR1P)=ALOG(PC2)
U(IR1P)=UC2
RH(IR1P)=RH(IR2)
S(IR1P)=S(IR2)
A(IR1P)=SQRT(GAM2*PC2/RHC2)
IDUM1=IR(1)
IDUM2=IR(2)
ZDUM1=(Z(IDUM1)-Z(2))/FLOAT(IDUM1-2)
ZDUM2=(Z(IDUM2)-Z(IDUM1))/FLOAT(IDUM2-IDUM1-1)
DZMIN=DZMIN*AMIN1(ZDUM1,ZDUM2)*0.999
DZMAX=DZMAX*DZMIN*1.001
RETURN
1 FORMAT(10E13.5)
2 FORMAT(14I5)
3 FORMAT(1I5,8E13.5)
END
```

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SUBROUTINE ACOUS(ZCON,ZBW,DT,CUNMD,RHORCT)

C\*\*\*\*\* VAR(1) IS TIME  
C\*\*\*\*\* VAR(2) IS Z  
C\*\*\*\*\* COMMON/F/RJET,XME,PE,AE,PINF,AINF,UINF,RHINF  
C\*\*\*\*\* COMMON/G/KK,LL,KD,TIME,TFINAL  
C\*\*\*\*\* COMMON/GA/GAM1,GAM2,CP1,CP2  
C\*\*\*\*\* COMMON/BLAST/IBW,BNCON,SPEED,RJET2  
C\*\*\*\*\* COMMON/KUTTA/ CUVAR(2),VAR(2),CI,DER(2),II  
DIMENSION ERRVAL(1),ELT(1),ELE1(1),ELE2(1)  
DIMENSION RFIT(20),PFIT(20),RHOFIT(20),RFITU(20),UFIT(20)  
DIMENSION EMFITP(20),EMFITR(20),EMFITU(20)  
DIMENSION XDUM(1),YDUM(1),DYDUM(1),D2YDUM(1)  
DATA IFIT, IDUM/20,1/  
DATA (RFIT(I),I=1,20)/-50.0,-45.0,-40.0,-35.0,-30.0,-27.5,-25.0,  
1 -22.5,-20.0,-17.5,-15.0,-12.5,-10.0,-7.5,  
2 -5.0,-2.5,0.0,2.5,5.0,7.5/  
DATA (PFIT(I),I=1,20)/0.0,-0.18,-0.24,-0.42,-0.62,-0.78,-0.90,  
1 -1.03,-1.20,-1.53,-1.90,-2.35,-2.83,-3.22,  
2 -3.40,-2.90,-0.60,4.30,13.0,23.1/  
DATA (RFITU(I),I=1,20)/-50.0,-42.5,-35.0,-27.5,-22.5,-17.5,-15.0,  
1 -12.5,-10.0,-7.5,-5.0,-4.0,-2.5,-1.0,0.50,  
2 2.0,3.5,4.5,6.0,7.5/  
DATA (UFIT(I),I=1,20)/0.0,0.04,0.095,0.162,0.22,0.30,0.35,0.41,  
1 0.52,0.65,0.77,0.825,0.91,0.97,1.0,0.955,  
2 0.755,0.55,0.045,-0.725/  
DATA HOL1/2H /,HOL2/2HBW/,HOL3/2HMD/,HOL4/2H C/  
EXTERNAL DERSUB,CHSUB  
DO 8 I=1,IFIT  
RHOFIT(I)=PFIT(I)/GAM2  
8 CONTINUE  
CALL SPLINE(RFIT,PFIT,IFIT,EMFITP)  
CALL SPLINE(RFIT,RHOFIT,IFIT,EMFITR)  
CALL SPLINE(RFITU,UFIT,IFIT,EMFITU)  
IMD=0  
N=1  
IEXT=0  
NT=1  
ELT(1)=TFINAL  
SPEC=0.0  
CI=DT  
CIMAX=0.0  
ELE1(1)=0.001  
ELE2(1)=0.0  
VAR(1)=TIME  
VAR(2)=ZBW  
II=0  
3 CONTINUE  
CALL INT1A(II,N,NT,CI,SPEC,CIMAX,IERR,VAR,CUVAR,DER,ELE1,ELE2,ELT,  
1ERRVAL,DERSUB,CHSUB,IEXT)  
KO=KO+1  
GO TO(1,1,2,2),IERR  
1 CONTINUE  
IF((KO.GE.KK,OR,VAR(1).GE.TFINAL) GO TO 5  
IF((KO/LL)=LL,NE,KO) GO TO 150  
5 CONTINUE

```

      WRITE(6,1000) KO,VAR(1)
      TDUM=VAR(1)
      ZDUM=VAR(2)
C*****CALCULATION OF THE MACH DISC LOCATION*****
C*****IF(IMD.EQ.1) GO TO 6
      IF(IMD.EQ.1) GO TO 6
      CALL MUZZLE(TDUM,PE,AE,XME)
      PRAT=PE/PINF
      IF(PRAT.GE.0.0) GO TO 7
      IMD=1
      GO TO 6
7 CONTINUE
      ZMD=CONMD*XME*SQRT(GAM1*PRAT)
      IF(ZMD.LE.1.0) IMD=1
6 CONTINUE
C*****CALCULATION OF THE ACOUSTIC WAVE PROPERTIES*****
C*****WRITE(6,4000)
      WRITE(6,4000)
      IF(IMD.EQ.0) WRITE(6,10) HOL3,ZMD
      HOL=HOL4
      DO 100 I=1,20
      RATIO=FLUA1(I-1)/19.0
      RDUM=ZCON+RATIO*(ZDUM-ZCON)
      XDUM(1)=(RDUM-SPEED*TDUM)/(RJET2 ALOG(RDUM/RJET2))
      IF(XDUM(1).GE.RFIT(1)) GO TO 20
      PRESS=1.0
      RHO=1.0
      UVAL=0.0
      GO TO 30
20 CONTINUE
      CALL SPLINT(RFIT,PFIT,IFIT,XDUM,1DUM,YDUM,DYDUM,D2YDUM,EMFITP)
      PRESS=YDUM(1)*RJET2/RDUM+1.0
      CALL SPLINT(RFIT,RHOFIT,IFIT,XDUM,1DUM,YDUM,DYDUM,D2YDUM,EMFITR)
      RHO=YDUM(1)*RJET2/RDUM+1.0
      CALL SPLINT(RFITU,UFIT,IFIT,XDUM,1DUM,YDUM,DYDUM,D2YDUM,EMFITU)
      UVAL=RHO-1.0-YDUM(1)*(RJET2/RDUM)**2*RHURCT
      30 CONTINUE
      IF(I.EQ.20) HOLE=HOL2
      ZDUMMY=RDUM/RJET2
      WRITE(6,10) HUL,ZDUMMY,PRESS,RHO,UVAL
      HOLE=HUL1
100 CONTINUE
      IF(KO.GE.KK,OR,VAR(1).GE.TFINAL) GO TO 4
150 CONTINUE
      GO TO(3,3,4,4),II
2 CONTINUE
      WRITE(6,2000) IEHR
      CALL EXIT
4 CONTINUE
      WRITE(6,3000)
      STOP
10 FORMAT(1X,A2,4E13.5)
1000 FORMAT(1H1,10X,*STEP =*,I4,10X,*TIME =*,E13.5,/)
2000 FORMAT(//,1X,26HINT1 NON-CONVERGENCE,IEHR=,I2)
3000 FORMAT(//,1X,17HEND CONDITION MET)

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4000 FORMAT(9X,3HZ/D,10X,1HP,12X,3HRHO,10X,1HU)  
END

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```

SUBROUTINE INTIA(I, N, NT, CI, SPEC, CIMAX, IERR, VAR, CUVAH, DER, ELE1,
  ELE2, ELT, ERRVAL, DEFSUB, CHSUB, ITEXT)
***** DOCUMENT DATE 03-01-68 SUBROUTINE REVISED 08-01-68 *****
DIMENSION SIVAR(20), SELE1(20), ELE1(20), ELE2(20), DER(21),
  FDERR(21), SDY(20), SDY1(20), VINCH(20), ERRVAL(20), ERVOVM(20),
  ZELT(10), SELT(13), KELMIN(20), STEP(3)
DIMENSION VAR(21), CUVAH(21)
INTEGER TEX(15)
INTEGER CODE, TPSH, SUMHAF, STEP, TEST, DCODE
REAL K1
C BEGIN INITIALIZATION
IF (I, GT, 0) GO TO 520
1 TP=0
  SSPEC=SIGN(SPEC,CI)
  SCIMAX=SIGN(CIMAX,CI)
  VAR1=  VAR(1)
  IF (CI .EQ. 0.0) GO TO 151
  IF (SSPEC .EQ. 0.0) GO TO 7
  IF (ABS(SCIMAX) .GT. ABS(SSPEC) .OR. SCIMAX .EQ. 0.0)
    ISCLMAX=SSPEC
C TEST TO SEE IF VAR IS ZERO
  IF(ABS(VAR1),GT.1,0E-11)GO TO 2
  TP=SSPEC
  GO TO 7
2 IF ((VAR1/SSPEC) .GT. 1.E-13) GO TO 4
3 K1=0.0
  GOTOB
4 K1=1.0
6 TP=VAR1-AMOD(VAR1,SSPEC)
  IF(ABS(TP-VAR1),LT.1.E-12)K1=1.0
  TP=TP+K1*SSPEC
  IF (ABS((TP-VAR1)/VAR1) .LT. 1.E-11) TP = TP + SPEC
C TEST FOR DIRECTION OF INTEGRATION
7 K1=1.0
  IF (CI .LT. 0.0) K1=-1.0
  CIK=CI*K1
  CIMAXK=SCIMAX*K1
  TPK=TP*K1
  VARK=VAR1*K1
C SET UP STORAGE FOR INTERNAL USE
  NP1=N+1
  NELT=1
  REMAIN=0.0
  NHAF=0
  NTSNT
  SUMHAF=0
  LOOP=0
  DO 91 I=1,3
91 STEP(I)=0
  IERR=1
  DO 8 I=1,NP1
8  CUVAH(I) =VAR(I)
  DO 101 I=1,N
101 SELE1(I)=ELE1(I)
  IF (NT .EQ. 0) GO TO 13
100 IF (NT ,EQ, 1) GO TO 10
  NTM1=NT-1

```

```

ELTK=K1*ELT(1)
DO 9 I=1,NTM1
ELTK2=K1*ELT(I+1)
IF (ELTK .LT. ELTK2) GO TO 9
GO TO 500
9 ELTK=ELTK2
10 CONTINUE
ELTK=K1*ELT(NELT)
IF (VARK .LT. ELTK) GO TO 11
IF (NELT .EQ. NT) GO TO 13
NELT=NELT+1
GO TO 10
11 NELT=NT-NELT+1
GO TO 12
13 NELTL=0
12 DO 14 I=1,N
14 RELMIN(I)=SELE1(I)/128.0
IF (NT .EQ. 0) GO TO 996
DO 995 I=1,NT
995 SELT(I)=ELT(I)
996 CALL DERSUB (II, CUVAR, DER, N, VAR)
IF (II .EQ. 4) GO TO 120
DO 15 I=1,N
15 FDERV(I)=DER(I+1)
II=1
TEST=0
DO 300 I=1,15
300 TEX(I)=0
TEX(1)=1
TEX(2)=1
KK3=1
IF (ITEXT) 635,63,635
151 WRITE (6,1000)
1000 FORMAT (//11H CI IS ZERO)
STOP
C END OF INITIALIZATION
520 II=1
TPSH=0
LTS=0
VARK=VAR(1)*K1
CIK=CI*K1
S1=YARK+CIK
IF (SSPEC ,EQ. 0.0) GO TO 525
KK=1
IF (NELTL, EQ. 0) GO TO 17
IF (ELTK-TPK) 16,16,17
16 CV=ELTK
CODE=1
GO TO 18
17 CV=TPK
CODE=2
18 IF(ABS(CV),LT,1,E-12)GO TO 530
IF (CV-S1)20,20,19
19 IF(ABS((CV-S1)/CV),GE,,1E-11)GO TO 535
20 IF (NELTL, EQ. 0) GO TO 540
IF(AHS((ELTK-TPK)/CV),LT,,1E-11)GO TO 550
IF (CODE ,EQ. 1) GO TO 545

```

540 DX=TP-VAR(1)

TEX(5)=1

TP=TP+SSPEC

TPK=TP\*K1

TPSH=1

GO TO 560

C SHORT INTERVAL DUE TO BOTH

550 TP=TP+SSPEC

TEX(6)=1

TPK=TP\*K1

TPSH=1

GO TO 545

C IF HERE CV IS LIKELY ZERO

530 IF(S1.LT.=-1.0E-12)GO TO 535

IF (CODE .EQ. 1) GU TO 550

IF (NELTL ,EQ, 0) GO TO 540

IF(ABS(ELTK),LT.1.0E-12)GU TO 550

GO TO 540

C SPEC IS ZERO

525 IF(ABS(REMAIN),GT.,1E-11)GO TO 96

IF (NELTL ,EQ, 0) GO TO 565

94 IF(ABS(ELTK),GE.1.E-12)GO TO 21

IF(S1.LT.=-1.0E-12)GO TO 565

GO TO 545

21 S2=ELTK-S1

IF(S2) 545,545,22

22 IF(ABS(S2/ELTK),LT.1.0E-12)GO TO 545

GO TO 565

C SHORT INTERVAL IS DUE TO ELT BLOCK

545 DELT= SELT(NELT)

TEX(4)=1

DX=DELT- VAR(1)

REMAIN=C1-DX

REMAIK=REMAIN\*K1

LTSHE=1

NELT=NELT+1

NELTL=NELTL-1

IF(NELTL,EQ,0)GO TO 560

ELTK=K1\*SELT(NELT)

GO TO 560

565 DX=C1

TEX(3)=1

GO TO 560

96 IF (NELTL ,EQ, 0) GO TO 98

IF (ELTK ,LT, (VARK+REMAIK)) GO TO 94

98 DX=REMAIN

TEX(7)=1

REMAIN=0,0

GO TO 560

535 DX=C1

TEX(3)=1

TEST=1

GO TO 555

C

C BEGIN RUNGE-KUTTA

C

560 TEST=0

```

555 DO 24 I=1,N
 24 S1VAR(I)= VAR(I+1)
575 CUVAR(1)=VAR(1)
576 DO 25 I=1,N
  SDY(I)=DER(I+1)
25 CUVAR(I+1)=S1VAR(I)+(DX*DER(I+1))/2.0
  CUVAR(1)=CUVAR(1)+DX/2.0
  CALL DERSUB (II,CUVAR,DER,N,VAR)
  IF (II .EQ. 4) GO TO 120
580 DO 26 I=1,N
  SDY(I)=SDY(I)+2.0*DER(I+1)
26 CUVAR(I+1)=S1VAR(I)+(DX*DER(I+1))/2.0
  CALL DERSUB (II,CUVAR,DER,N,VAR)
  IF (II .EQ. 4) GO TO 120
585 DO 27 I=1,N
  SDY(I)=SDY(I)+2.0*DER(I+1)
27 CUVAR(I+1)=S1VAR(I)+DX*DER(I+1)
  CUVAR(1)=CUVAR(1)+DX/2.0
  CALL DERSUB (II,CUVAR,DER,N,VAR)
  IF (II .EQ. 4) GO TO 120
590 DO 90 I=1,N
  SDY(I)=(SDY(I)+DER(I+1))/6.0
90 CONTINUE
  IF (LOOP) 28,28,29
28 DO30 I=1,N
  SDY1(I)=SDY(I)
  YINCR(I)=0.0
30 DER(I+1)=FDERV(I)
  DX=DX/2.0
  LOOP=1
  GO TO 575
C
C   LOOP WAS NOT ZERO
C
29 DO 31 I=1,N
31 YINCR(I)=YINCR(I)+SDY(I)
  IF (LOOP ,EQ, 2) GO TO 33
  DO 32 I=1,N
  S1VAR(I)=VAR(I+1)+DX*YINCR(I)
32 CUVAR(I+1)=S1VAR(I)
  CUVAR(1)=VAR(1)+DX
  LOOP=2
  CALL DERSUB (II,CUVAR,DER,N,VAR)
  IF (II .EQ. 4) GO TO 120
  GO TO 576
33 LOOP=0
  H=2.0*DX
  DO 34 I=1,N
  ERVOVM(I)=(YINCR(I)/2.0-SDY1(I))/15.0
  ERRVAL(I)=H*ERVOVM(I)
34 S1VAR(I)=S1VAR(I)+DX*SDY(I)+ERRVAL(I)
C
C   S1VAR HOLD THE APPROXIMATE ANSWERS
C
  IF (SCIMAX) 36,35,36
36 IF(ABS(SCIMAX-CI),LT,1.0E-12)GO TO 38
35 IF(ABS(H-CI),GT,1.0E-12)GO TO 38

```

```

DCODE=0
GO TO 605
38 DCODE=1
605 CONTINUE
I=0
40 I=I+1
IF (I .GT. N) GO TO 45
IF (ABS(S1VAR(I)) .LT. ELE2(I)) GO TO 40
RELER=AH(SERRVAL(I)/S1VAR(I))
IF (RELER .GT. SELE1(I)) GO TO 615
IF (RELER .GT. RELMIN(I)) DCODE=1
GO TO 40
45 CONTINUE
IF (DCODE=1) 610,620,610
610 CONTINUE
IF (SSPEC)41,42,41
42 IF (SCIMAX)41,43,41
43 CI=2.0*CI
TEX(8)=1
NHAFF=NHAFF+1
GO TO 620
41 IF (2.0*ABS(CI) .LE. ABS(SCIMAX)) GO TO 43
44 CI=SCIMAX
TEX(8)=1
GO TO 620
C
C HALF INTERVAL
615 NHAFF=NHAFF+1
TEX(9)=1
NVAR=I+1
IF (NHAFF=8)47,47,505
47 IF (LTSH .EQ. 0) GO TO 48
TEST=1
LTSH<0
NLT=NLT-1
NLT=NLT+1
ELTK=K1*SELT(NLT)
REMAIN=0.0
48 IF (TPSH .EQ. 0) GO TO 49
TEST=1
TP=TP-SSPEC
TPK=K1*TP
TPSH<0
49 IF (SSPEC .NE. 0.0) GO TO 998
TEST=0
IF (ABS(CI-2.0*DX) .GT. 1.E-12) GO TO 1100
998 CI=DX
999 UX=DX/2.0
CIK=K1*CI
DO 50 I=1,N
S1VAR(I)=VAR(I+1)
DER(I+1)=FDERV(I)
SDY1(I)=YINCR(I)-SDY(I)
50 YINCR(I)=0.0
KK3=2
IF (ITEXT .EQ. 1) GO TO 637
99 LOOP=1

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GO TO 575  
 1100 CONTINUE  
 IF (NHA<sub>F</sub> .GT. 1) GO TO 999  
 NTS=NTS+1  
 IF (NTS .GT. 13) GO TO 998  
 ACV=VAR(1)+CI  
 ACVK=ACV\*K1  
 IF (NELTL .EQ. 0) GO TO 1102  
 NLT=NELT  
 1101 ELTK1=SELT(NLT)\*K1  
 IF (ACVK .LT. ELTK1) GO TO 1103  
 NLT=NLT+1  
 IF (NLT .EQ. NTS) GO TO 1106  
 GO TO 1101  
 1102 SELT(NELT)=ACV  
 GO TO 1105  
 1103 NLTP1=NLT+1  
 I=NTS  
 1108 SELT(I)=SELT(I-1)  
 IF (I .EQ. NLTP1) GO TO 1106  
 I=I-1  
 GO TO 1108  
 1106 SELT(NLT)=ACV  
 1105 NELTL=NELTL+1  
 TEX(9)=0  
 TEX(10)=1  
 ELTK=K1\*SELT(NELT)  
 GO TO 999

C  
 C DOUBLE PRECISION UPDATING  
 C

---

620 LOOP=0  
 DH=H  
 DO 51 I=1,N  
 PHI=ERVOVM(I)+YINCR(I)/2.0  
 DPHI=PHI

---

51 CUVAR(I+1)=VAR(I+1)+DH\*DPHI  
 CUVAR(1)=VAR(1)+DH  
 CALL DEHSUB(I1,CUVAR,DER,N,VAR)  
 IF (II .EQ. 4) GO TO 120  
 CALL CHSUB(II,C1,VAR,CUVAR,DER)  
 IF (II-2) 54,600,121

---

121 TEST=0  
 54 DO 57 I=1,N  
 57 FDERV(I)=DER(I+1)  
 SUMHAF=SUMHAF+NHA<sub>F</sub>-STEP(1)  
 STEP(1)=STEP(2)  
 STEP(2)=STEP(3)  
 STEP(3)=NHA<sub>F</sub>  
 NHA<sub>F</sub>=0  
 IERR=1  
 IF (SUMHAF=8) 63,63,510

---

63 DO 59 I=1,NP1  
 59 VAR(I)=CUVAR(I)  
 TEX(12)=1

---

501 KK3=4  
 IF (1TEXT .EQ. 1) GO TO 637

58 IF (TEST .EQ. 1) GO TO 520  
 120 RETURN

C

C RECOMPUTE INTERVAL

C

600 TEST=0

NMAF=0

II=1

DX=CI

TEX(1)=1

KK3=3

IF (ITEXT .EQ. 1) GO TO 636

70 CIK=CI\*K1

DO 60 I=1,N

DER(I+1)=FDERV(I)

60 CUVAR(I)= VAR(I)

CUVAR(N+1)= VAR(N+1)

IF (TPSM .EQ. 0) GO TO 61

TP=TP-SPEC

TPK=TP\*K1

TPSM=0

61 IF (LTSH .EQ. 0) GO TO 555

NELT=NELT-1

REMAINE=0.0

NELTL=NELTL+1

ELTK=SELTI(NELT)\*K1

GO TO 555

636 WRITE(6,183) VAR(1),DX

GO TO 102

635 IF(TEX(1).EQ.1) WRITE(6,171) VAR(1)

IF(TEX(2).EQ.1) WRITE(6,172) CI,CIMAX,SPEC

637 IF(TEX(3).EQ.1) WRITE(6,173)

IF(TEX(4).EQ.1) WRITE(6,174) H

IF(TEX(5).EQ.1) WRITE(6,175) H

IF(TEX(6).EQ.1) WRITE(6,176) H

IF(TEX(7).EQ.1) WRITE(6,184) H

IF(TEX(8).EQ.1) WRITE(6,177) CI

IF(TEX(9).EQ.1) WRITE(6,178) NVAR, CI

IF(TEX(10).EQ.1) WRITE(6,185) NVAR, DX

IF(TEX(11).EQ.1) WRITE(6,183) VAR(1), DX

IF(TEX(12).EQ.1) WRITE(6,179) VAR(1)

IF(TEX(13).EQ.1) WRITE(6,180)

IF(TEX(14).EQ.1) WRITE(6,181)

IF(TEX(15).EQ.1) WRITE(6,182)

102 DO 320 I=3,13

320 TEX(I)=0

GO TO (120,99,70,58),KK3

171 FORMAT (33H INITIALIZATION STARTS AT VAR(1)=,E16.8/)

172 FORMAT (4H CI=,E15.8,9H CIMAX=,E15.8,8H SPEC=,E15.8/)

173 FORMAT (37H DX IS THE FULL COMPUTING INTERVAL CI/)

174 FORMAT (28H DX IS A SHORTENED INTERVAL ,E15.8,25H DUE TO A CRITICAL VALUE/)

175 FORMAT (28H DX IS A SHORTENED INTERVAL ,E15.8,21H DUE TO A SPECIFIC VALUE/)

176 FORMAT (28H DX IS A SHORTENED INTERVAL ,E15.8,39H DUE TO BOTH A SPECIFIC AND CRITICAL VALUE/)

177 FORMAT (27H CI HAS BEEN LENGTHENED TO ,E16.8/)

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178 FORMAT (5H VAR(,12,32H) HAS CAUSED CI TO BE HALVED TO ,E16.8/)  
179 FORMAT (27H VAR(1) HAS BEEN UPDATED TO ,E16.8,1)  
180 FORMAT (31H ERROR RETURN-ELT NOT MONOTONIC/)  
181 FORMAT (55H ERROR RETURN-HAVE HALVED 9 TIMES OVER LAST 3 INTERVALS  
1/)  
182 FORMAT (45H ERROR RETURN-HAVE HALVED 9 CONSECUTIVE TIMES/)  
183 FORMAT (31H INTERVAL RECOMPUTED AT VAR(1)=,E16.8,9H WITH DX=,E16.8  
1/)  
184 FORMAT (25H DX IS SHORTENED INTERVAL,E16.8,28H DUE TO A PREVIOUS EL  
1T VALUE/)  
185 FORMAT (5H VAR(,12,32H) HAS CAUSED DX TO BE HALVED TO ,E16.8,38H BU  
1T NOT CI SINCE CI ALREADY SHORTENED/)  
500 IERR=2  
TEX(13)=1  
TEST=0  
GO TO 63  
505 IERR=3  
TEX(15)=1  
TEST=0  
GO TO 501  
510 IERR=4  
TEST=0  
TEX(14)=1  
GO TO 63  
END

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```
SUBROUTINE DERSUB
COMMON/KUTTA/ CUVAR(2),VAR(2),CI,DER(2),II
COMMON/F/RJET,XME,PE,AE,PINF,AINF,UINF,RHINF
COMMON/BLAST/IBW,BWCON,SPEED,RJET2
RDIST=CUVAR(2)/RJET2
DER(2)=AINF/SQRT(1.0-BWCON/(SQRT(ALUG(RDIST))*RDIST))
RETURN
END
```

SUBROUTINE CHSUB  
RETURN  
END

TM 184  
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